Cryogenic DT and D₂ Targets for Inertial Confinement Fusion



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Summary

Ignition-scaled cryogenic DT capsules are being imploded on the OMEGA Laser System

- The success of the cryogenic target handling system at LLE is the culmination of three decades of research and development.
- Since 2001, 118 cryogenic D_2 and 15 cryogenic DT capsules have been imploded on the OMEGA laser.
- β -layering produces inner-ice smoothness well below the ~1- μ m-rms (all modes) requirement for direct- and indirect-drive ignition.
- IR layering of D₂ targets produces similar results.
- The NIF cryogenic target development program has produced and characterized ignition-quality ice layers in Be shells.

Peak fuel areal densities approaching 200 mg/cm² have been achieved in both D_2 and DT implosions.

A goal of ICF research is to demonstrate ignition and burn at the lowest possible incident energy



Ablation is used to generate the extreme pressures required to compress a fusion capsule to ignition conditions



The basic capsule design and fabrication tolerances are similar for both direct- and x-ray-drive ignition



Perturbation seeds early in the implosion determine the final capsule performance



The inner-ice-smoothness requirements are similar for direct- and x-ray-drive ignition



¹P. W. McKenty *et al.*, Phys. Plasmas <u>8</u>, 2315 (2001).

²J. D. Lindl et al., Phys. Plasmas <u>11</u>, 339 (2004).

The challenge for ignition on the NIF is to fabricate the x-ray-drive point design target to specifications

Outer-surface finish is close to specifications.

Polished Be capsule



Minimal impact on performance.

UR LLE

DT layer in Be capsule

Ice-surface smoothness is close to specification.

DT ice surface

Cryogenic hohlraum



Low-mode isotherm control demonstrated.

Outline



- Brief history
- Making a smooth D₂/DT ice layer
- Characterizing the layer smoothness
- Imploding the target
- Results of recent cryogenic DT implosions on OMEGA

Inertial confinement fusion ignition and gain is made possible by two technological breakthroughs

- 1. β -layering of the DT fuel
- 2. Inner-ice-surface characterization using optical shadowgraphy (transparent ablators) and x-ray phase-contrast imaging (opaque ablators).

Cryogenic target development evolved rapidly following pioneering research at KMS, Illinois, LLNL, LANL, and LLE

巨狗

KMS shoots thin cryo DT-glass
shells using point-contact
conduction cooling $(Y_n \sim 10^{6-7})$
(Henderson & Johnson, 1977)Illinois/KMS use cold He gas
to fast refreeze thin DT layers
for improved layer uniformity
(Kim & Reiger, 1980)1977 \longrightarrow 1979 \longrightarrow 1980KMS uses a fast shroud-retraction
scheme to expose target at shot time

(Mulsinski et al., 1979)

Cryogenic target development evolved rapidly following pioneering research at KMS, Illinois, LLNL, LANL, and LLE

LLE



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The first step in DT/D₂ layering is to establish an isothermal temperature near the triple point on the capsule surface



- Low-pressure He gas is used to exchange heat with an external "isothermal boundary."
- dT/dr across the ice becomes positive by depositing heat, Q, directly into the bulk ice.
- If *dT/dr* across the ice is > 0, thicker (warmer) regions sublimate and deposit on thinner (cooler) regions.

$$C_{v}dT/dt + \nabla(\kappa_{ice}\nabla T) = Q = -\kappa_{He}dT/dr$$

$$T_{in} = T_{out} + Qh^{2}/2\kappa_{ice} \text{ (typically <1 mK)}$$

A spherically symmetric temperature gradient across the DT or D_2 ice will form a uniform ice layer.

Heat can be deposited into the bulk ice by the radioactive decay of tritium¹ or absorption of IR radiation²



¹J. K. Hoffer and L. R. Foreman, Phys. Rev. Lett. <u>60</u>, 1310 (1988). ²G. W. Collins *et al.*, J. Vac. Sci. Technol. A <u>14</u>, 2897 (1996).

Physically mounting the capsule affects the low-mode symmetry of the isotherm

A robust mechanical support is required to ensure target survival and precision alignment. **Direct-drive target Indirect-drive target** Low pressur He → Coolina $\nabla T \rightarrow$ Capsule (**Be**) Fill tube Low-→ Coolina pressure helium Isotherm established Isotherm established by cylindrical by spherical cavity hohlraum IR illumination imposes lowmode asymmetry by preferentially

Low-mode thermal symmetry must be actively controlled in a hohlraum

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heating the target structures

A 3-D model is essential to understand the trade-offs between structural integrity and thermal uniformity



For direct-drive targets, thermal symmetry is achieved with clever design rather than active controls

For transparent ablators, the smoothness of the inner ice surface is measured using optical shadowgraphy



D. H. Edgell, session JO2.00004

A 3-D representation of the inner ice surface can be constructed from multiple views (typically 48)



Structures in the ice correlate with known asymmetries in the thermal environment and are consistent over repeated layering/melting cycles.

For input to simulations, the Legendre modes can be mapped from the average Fourier components^{*}



- A direct fit to the 3-D surface can be used for $\ell \lesssim 12$
- 2-D Fourier components (*P_n*) are averaged over many great circles
- Smoothed (*P_n*) to improve mapping behavior
- $P_{\ell} = \sum_{n=\ell, \ell+2, \dots}^{\infty} a_{\ell n} \langle P_n \rangle$
- Assumes isotropic distribution of perturbations

A more-accurate spectrum is produced by starting with a large number of independent views.

*S. Pollaine and S. Hatchett, Nucl. Fusion <u>44</u>, 117 (2004).

X-ray-phase contrast imaging* is used to characterize the ice in opaque shells (e.g., Be, C, and foam)



Need sufficient spatial coherence (plane wave or point source) to obtain good contrast for a given imaging configuration.

X-ray-phase contrast images reveal detailed structures in Be(Cu) DT cryogenic capsules



Gradual solidification virtually eliminates high-spatial-frequency surface roughness



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"Single-crystal" ice layers grown at the triple point require temperature control to mK precision

3-D thermal modeling led to a significant improvement in ice smoothness for D₂ capsules (IR layering)

- Layering-sphere upgrades based on thermal modeling
 - Increased the exchange-gas pressure
 - Added a diffuser to the IR laser-fiber input
 - Modified the mount structures to minimize IR absorption (e.g., Au coating)





LLE

Several DT (45:55) targets with an ice roughness of \lesssim 1- μ m rms for all modes have been imploded on OMEGA



More than half of the DT capsules created to date have produced layers with sub-1- μ m-rms roughness



On the NIF, DT will be introduced into the Be(Cu) shell through a fill tube



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Ice-layer thickness can be controlled with high accuracy.

DT layers in Be at 0.3 mg/cc meet the NIF smoothness standard for modes ≥ 10



• Modes 1 to 3 add about 2 μ m to the rms value

LLNL has successfully shimmed the axial P1 mode in an aluminum hohlraum



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Excessive exposure to ambient chamber radiation at shot time will significantly affect the layer quality



The modeling indicates that exposure times of less than 100 ms are required to preserve the layer for direct drive.

The rapid removal of a thermally passive shroud meets the requirements for exposure time and thermal uniformity

OMEGA-24 shroud and target assembly (1987–1988)



Based on original KMS concept¹

- Opposed port shroud retractor using a linear induction motor (~10-ms exposure to chamber)
- Insertion/retraction mechanisms decoupled from the target chamber
- He gas cooling for highuniformity, fast-refreeze, thin-DT-glass-shell targets (5-μm-thick DT and 3- to 7-μm-thick glass)
- Interferometric layer characterization

¹D. L. Musinski *et al.*, Appl. Phys. Lett. <u>34</u>, 300 (1979).

The OMEGA-24 CTHS performed over 100 cryogenic DT target shots and produced 200× DT liquid density¹



relevant target designs (e.g., thick DT layers and CH ablators).

The conceptual design of the OMEGA-60 CTHS has roots in the earlier systems

New requirements included:

- high-pressure permeation fill
- up to 100-µm-thick ice
- fill 12 targets per week
- IR-enhanced layering for D₂
- optical characterization
- moving cryostat
- alignment to 5 μ m relative to TCC
- exposure time <100 ms
- vertical shroud pull



Cryostat assembly is for target positioning and life support.

Design was started in 1992 at General Atomics and was completed in 1999; first cryogenic D₂ implosion in 2000!

LLE typically implodes 2 to 4 cryogenic capsules per day, two days per month (DT and D_2)



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The key to the sucess of the OMEGA-60 CTHS is the moving cryostat



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On the NIF, cryogenic DT targets will be filled and characterized on the chamber



- target loading glovebox
- x-ray imaging for fuel-ice-layer characterization

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X-Ray Drive Cryogenic hohlraum targets on the NIF will be fielded using a "clam-shell" retractor



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Hot-spot physics can be done on OMEGA with ignition-quality ice

- Fuel assembly (areal density)/convergence
 - P. B. Radha, GO2.00008
 - J. A. Frenje, GO2.00009 and JO2.00003

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- D. D. Meyerhofer, ZO1.00001
- Core temperature
- Yields
 - V. Yu. Glebov, GO2.00011
- Laser-energy coupling
 - W. Seka, ZO1.00002
 - I. V. Igumenshchev, ZO1.00013
 - V. N. Goncharov, ZO1.00012
 - S. P. Regan, ZO1.00005
- Performance sensitivity to single beam smoothing
- Performance sensitivity to adiabat shaping
 P. W. McKenty, VO2.00002
- Performance sensitivity to shell stability (adiabat)

15 DT implosions and 118 D₂ implosions have been performed on OMEGA

The areal density increases significantly during the fusion burn

Shot 40857, CH[27]D₂(15), $\alpha = 2$ 250 Observed flux (arbitrary units) $\rho \mathbf{R}_{\mathrm{peak}}$ 200 $\langle
ho {\sf R}
angle_{\sf n}$ 150 mg/cm² **Neutron** rate 100 X ray 50 (2 keV) 0 3 5 4 Time (ns)

The neutron-averaged areal density $\langle \rho R \rangle_n$ is greater than 100 mg/cm² for cryogenic D₂ implosions



^{*}V. A. Smalyuk et al., Phys. Rev. Lett. <u>90</u>, 135002 (2003).

The peak areal density ρR_{peak} may be inferred by using core self emission to backlight the fuel shell

Emitted x-ray spectrum is 1-D simulations can be used to estimate the product of a source term ho and suggest the $ho {\sf R}_{\sf peak}$ could be and an attenuation term as high as 180 to 190 mg/cm² Bremsstrahlung D₂ cryogenic implosion Fluence (10¹⁶ keV/keV) 10 Hot **Absorption** $e^{-E/T}$ X rays spot elated to ρR at T_o ħω Cold shell • Spect = $(e^{-E/kT}_{hot}) \times (e^{-\mu\rho R}_{shell})$, where μ is the mass attenuation coefficient and is proportional to ρ Experiment LILAC LILAC, opacity = 0 The fuel-shell attenuation 0.1 is proportional to $\rho^2 R$ 1.5 2.5 2.0 3.0 3.5 Photon energy (keV)

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2-D simulations are expected shortly to confirm fuel density estimates

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