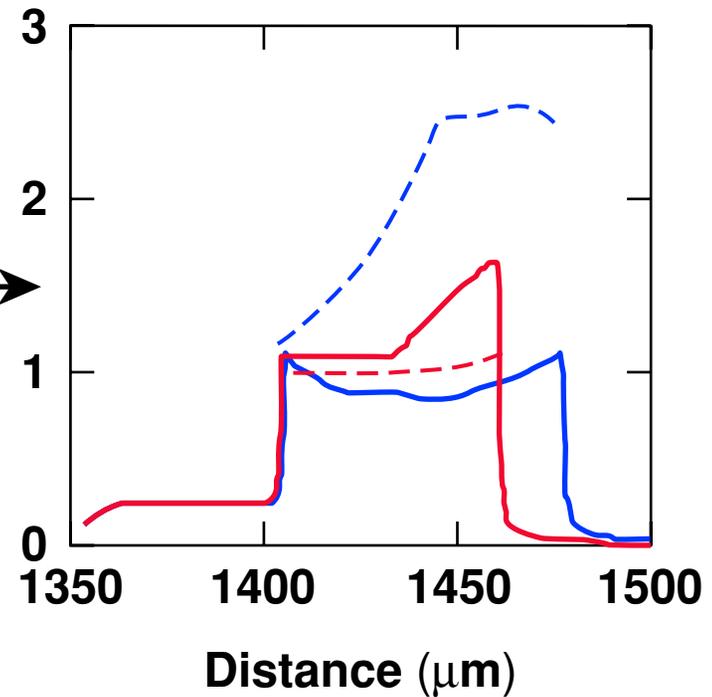
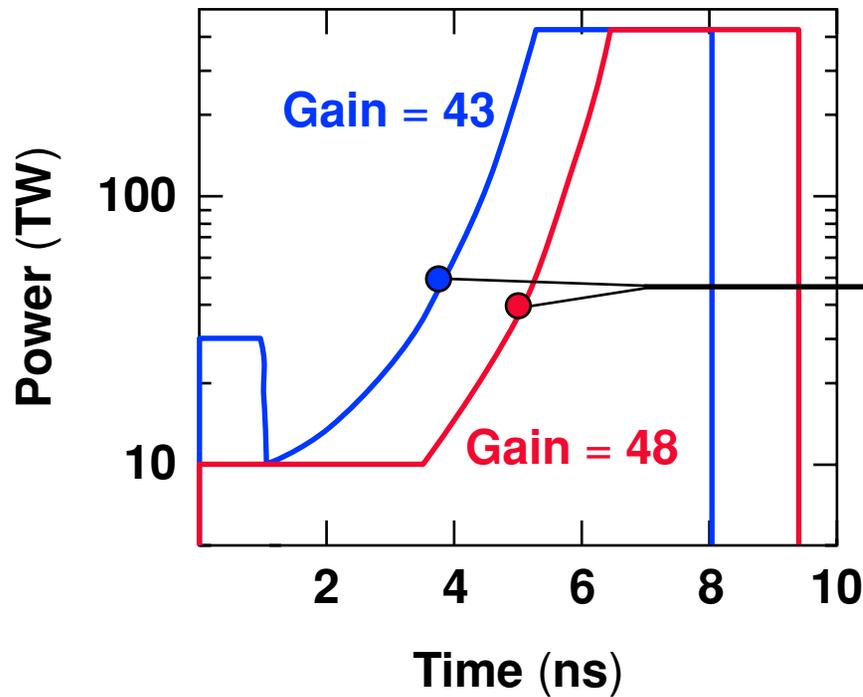


# Optimization of Direct-Drive Target Designs for the NIF



— Density ( $\text{g}/\text{cm}^3$ )    -- Adiat/3

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American Physical Society  
Division of Plasma Physics  
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## Summary

# A model has been developed to optimize NIF DD target designs

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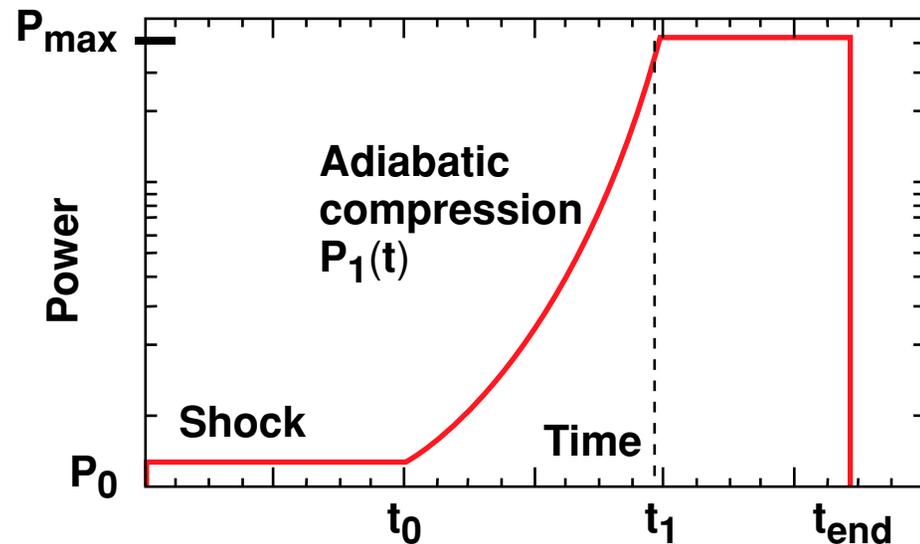
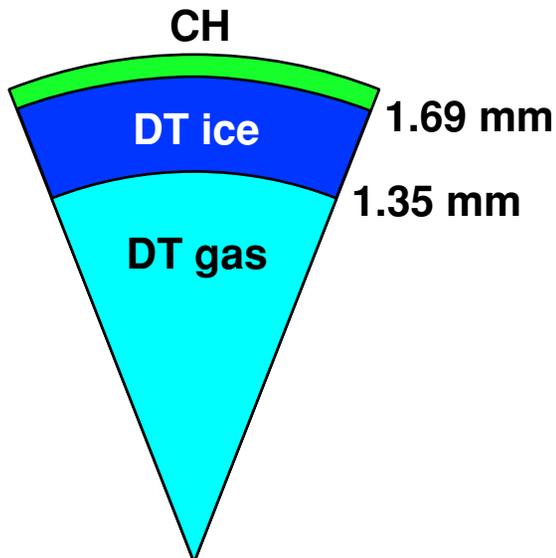


- A model has been developed to optimize target gain.
  - The model uses results of a stability postprocessor to calculate shell integrity during the acceleration phase and mode spectrum at shell stagnation.
  - Target gain is calculated by using the obtained mode spectrum and results of 1-D simulations with reduced implosion velocities.
- The model was applied to predict stability and gains for “all-DT” moderate-gain and high-gain foam target designs.
- The results of the model suggest that the maximum gain for the “all-DT” targets can be achieved for  $\alpha = 3$  to  $\alpha = 4$  designs.

# The model consists of three main steps

<b>Step</b>	<b>Instability seeding</b>	<b>Acceleration, coasting, and deceleration phases</b>	<b>Target gain</b>
<b>Physical phenomena</b>	<b>Imprinting, RMI, feedout</b>	<b>Ablative RTI, Bell-Plesset instability</b>	<b>Burn-wave propagation</b>
<b>Calculated by</b>	<b>Analytic theory, <i>ORCHID</i> simulations</b>	<b>Stability postprocessor</b>	<b>1-D simulations with reduced implosion velocity</b>

# “All-DT” DD NIF targets driven on adiabat up to 7 were considered

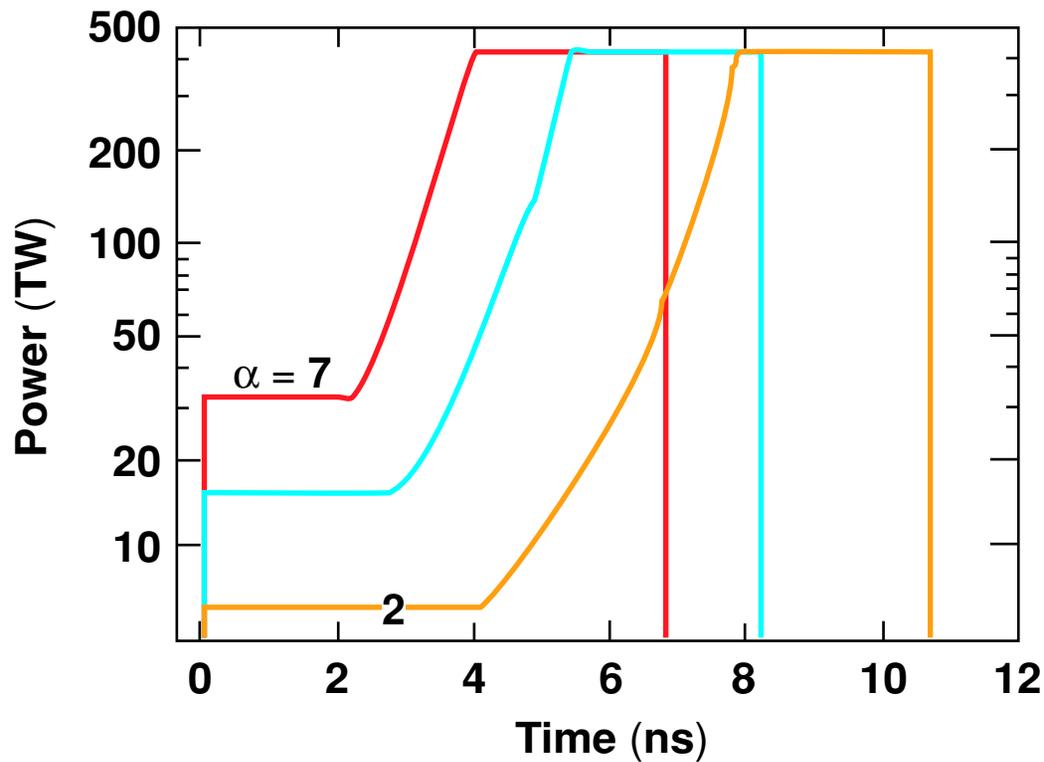


$$\text{Pressure } p_{\text{kidder}} = \frac{p_0}{\left[1 - (t/\tau)^2\right]^{5/2}}$$

$$\text{Power } P_1 = \frac{P_0}{\left[1 - (t/\tau)^2\right]^4}$$

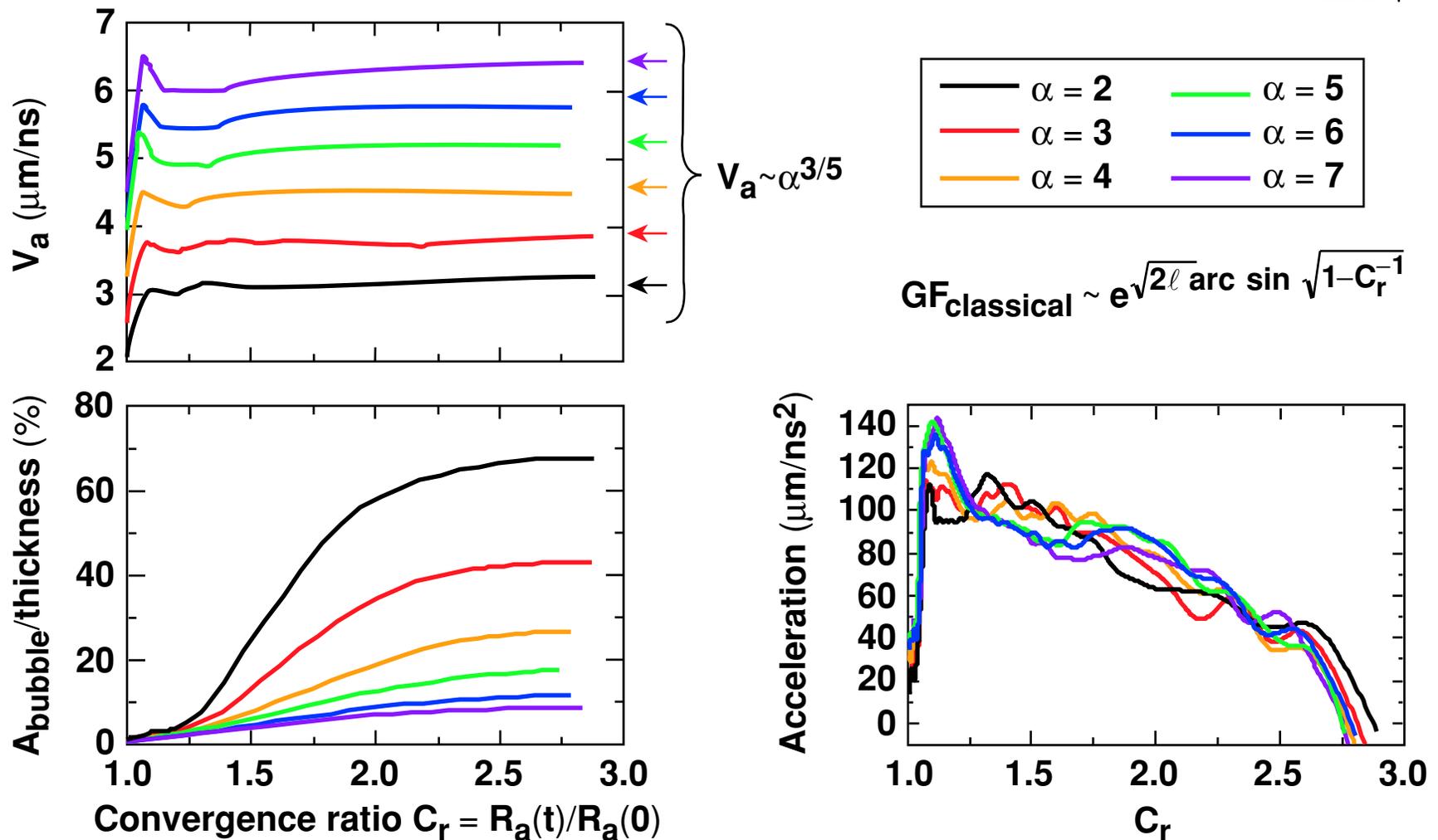
Parameter	Determined by
$P_0$	Shell adiabat
$P_{\text{max}}$	Damage threshold, shell stability
$t_0$	Timing of compression wave and first shock
$t_1$	Target gain
$t_{\text{end}}$	Laser energy

# “All-DT” DD NIF targets driven on adiabat up to 7 were considered (continued)



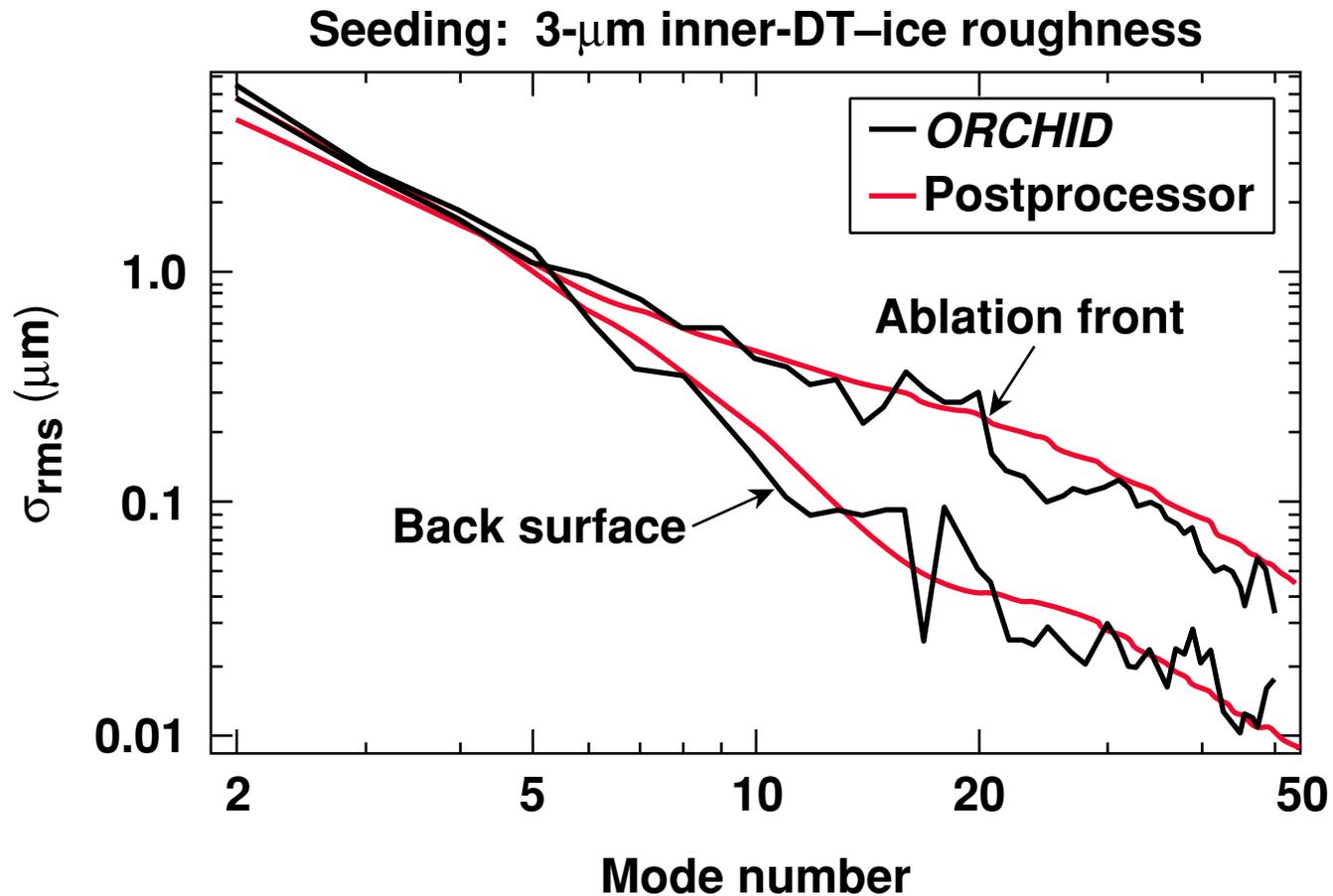
$\alpha$	$\rho R_{\text{peak}}$ (g/cm <sup>2</sup> )	$V_{\text{imp}}$ ( $\times 10^7$ cm/s)	Gain
2	1.5	4.17	55
3	1.3	4.27	48
4	1.2	4.34	41
5	1.1	4.42	29
6	1.0	4.42	22
7	0.9	4.45	9

# A stability postprocessor<sup>1</sup> was applied to study perturbation evolution of imploding targets during the acceleration, coasting, and deceleration phases



# Result of the model was compared against *ORCHID* simulations

- End of acceleration phase

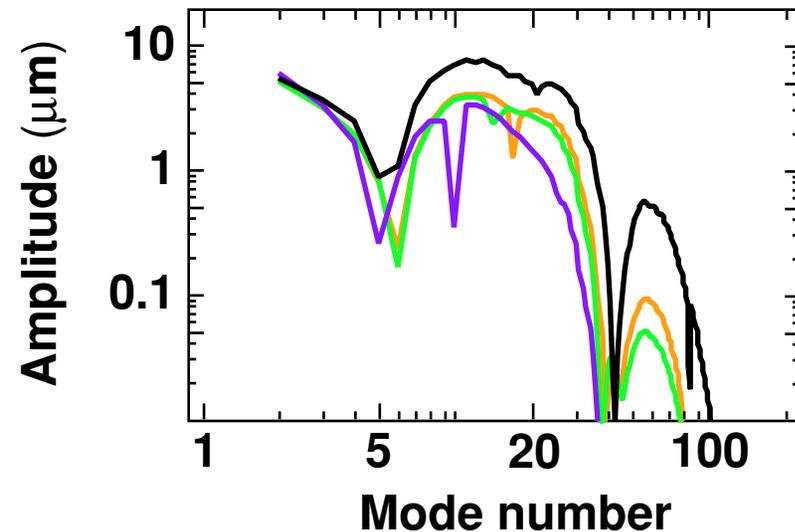
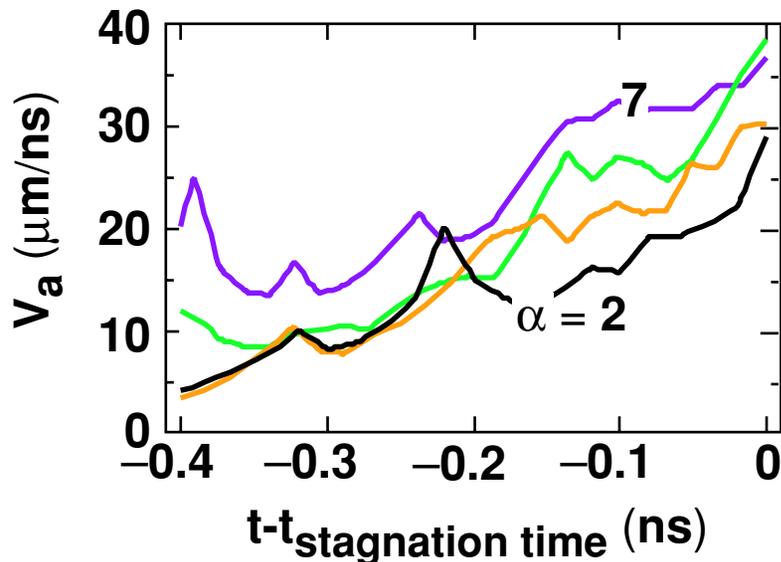


# The postprocessor was used to calculate mode spectrum at stagnation

- $V_a$  in decel phase is calculated by using theory of R. Betti<sup>1</sup> *et al.*

$$V_a \propto \frac{(T_{hs})^{5/2}}{R_{hs} \rho_{shell}}$$

- Mode spectrum at the back surface of cold fuel at stagnation (1 THz SSD, 1  $\mu\text{m}$  DT ice roughness, 800 $\text{\AA}$  outer surface finish)



<sup>1</sup>V. Lobatchev and R. Betti, Phys. Rev. Lett., 85, 4522 (2000).

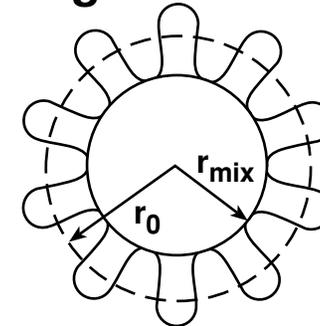
# The mode spectrum at stagnation is related to the gain reduction

- According to Levedahl and Lindl<sup>1</sup>

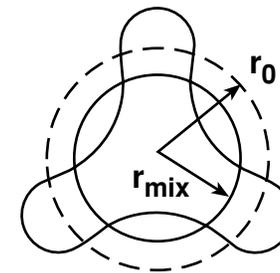
$$\frac{V_{\text{mix}}}{V_0} = \left( \frac{r_0}{r_{\text{mix}}} \right)^{2/5} \quad r_{\text{mix}} = r_0 - \eta$$

$$\frac{V_{\text{mix}}}{V_0} = (1 - \xi)^{-2/5} \quad \xi = \eta/r_0$$

High- $l$  modes



Low- $l$  modes



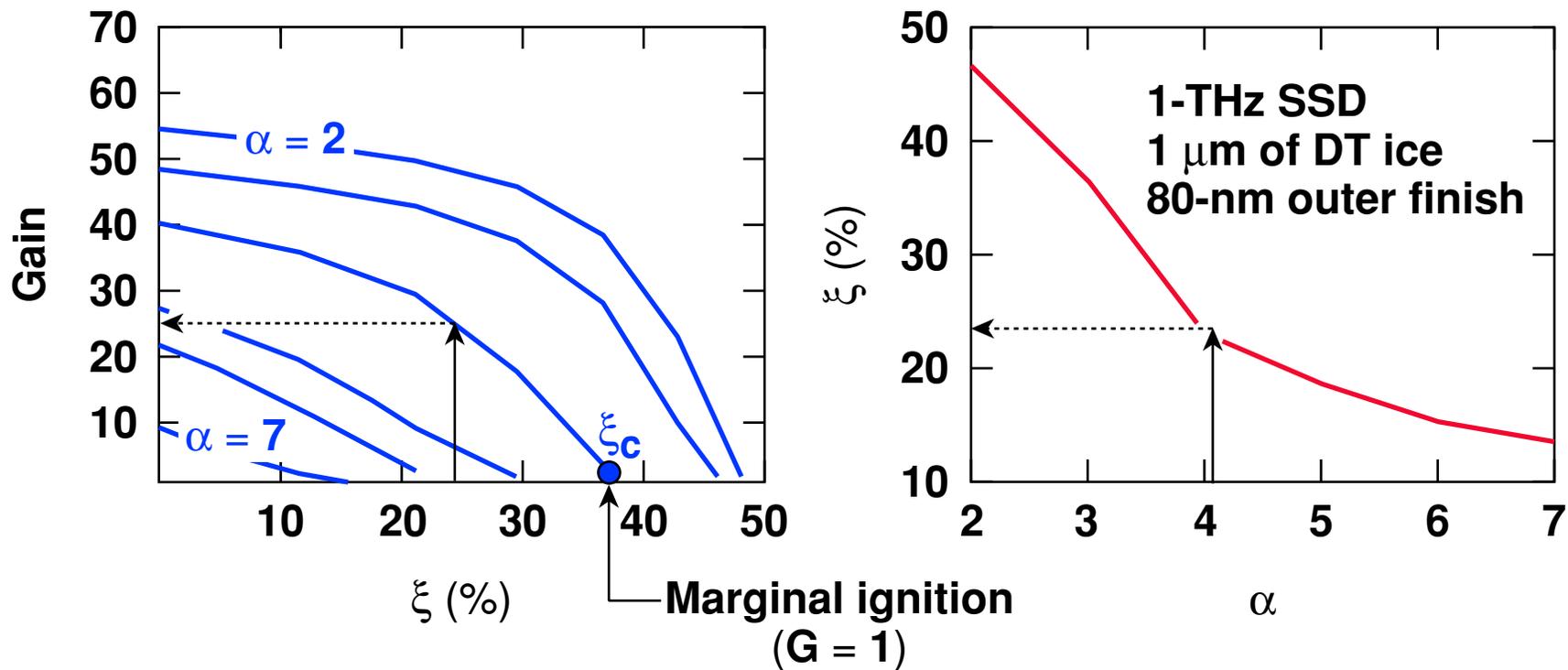
- Perturbation is equivalent to a reduction in 1-D implosion velocity:<sup>2</sup>

$$\xi = 1 - \left( \frac{V_0 - \Delta V}{V_0} \right)^{5/2}$$

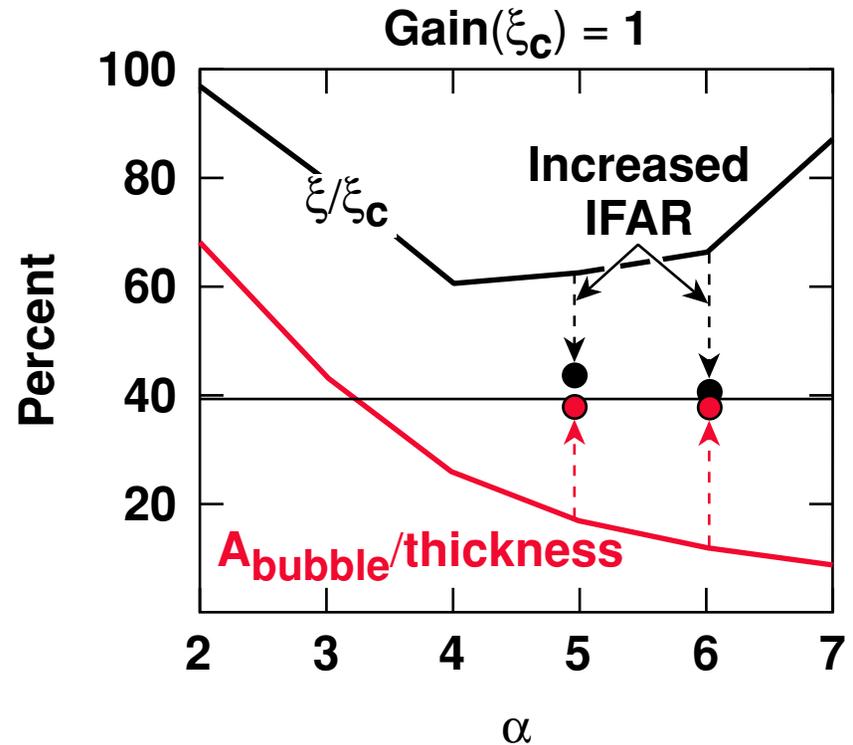
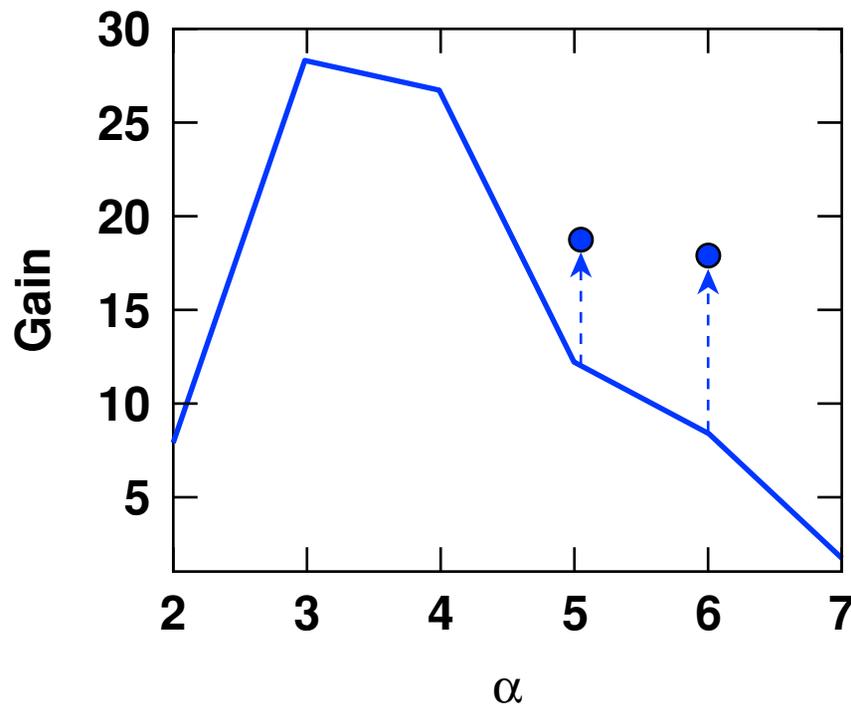
<sup>1</sup>W. Levedahl and J. Lindl, Nuc. Fusion 37, 165 (1997).

<sup>2</sup>Roy Kishony, Ph.D. thesis, 1999.

Gain is calculated by using the results of 1-D simulations with reduced  $V_{imp}$



# Target designs with $3 < \alpha < 4$ have the highest 2-D gain



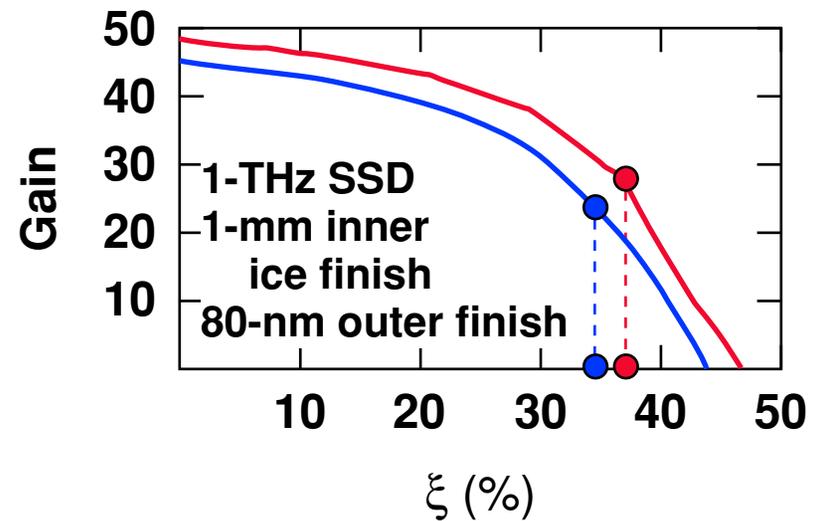
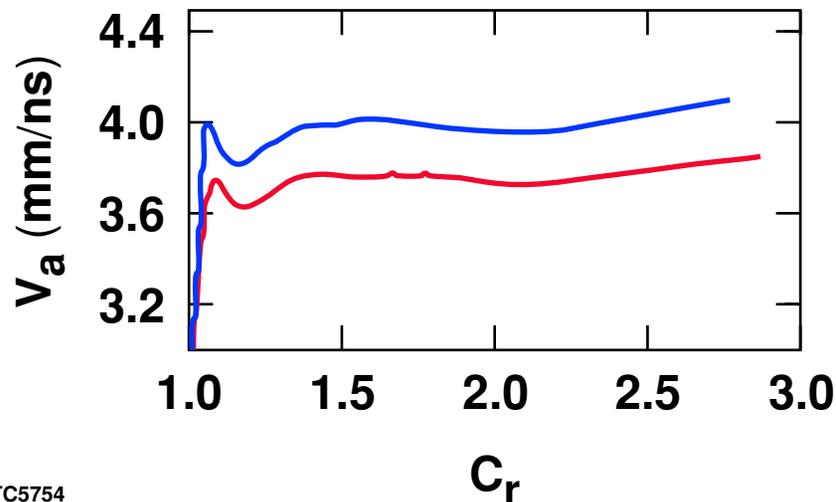
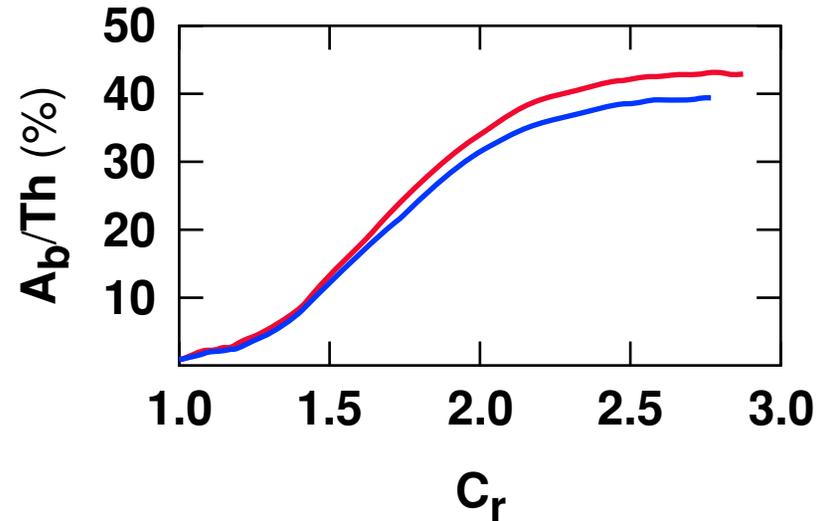
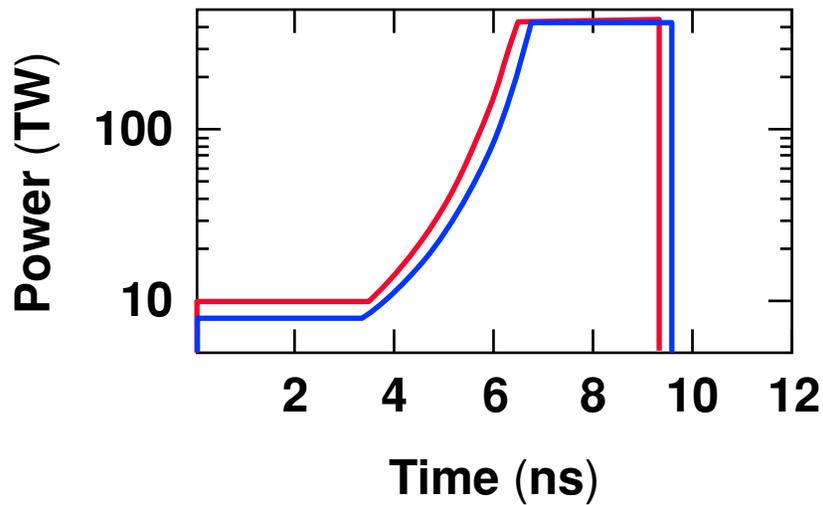
- Target gain is calculated assuming 1-THz, 2-D SSD; 1- $\mu\text{m}$  ice-DT gas roughness, and 800- $\text{\AA}$  outer surface finish.
- Imprint spectrum is assumed to be the same for different  $\alpha$ 's.

# Change in EOS results in a small variation in target gain

$\alpha = 3$  design

*SESAME* ( $G = 48$ )

TF ( $G = 45$ )



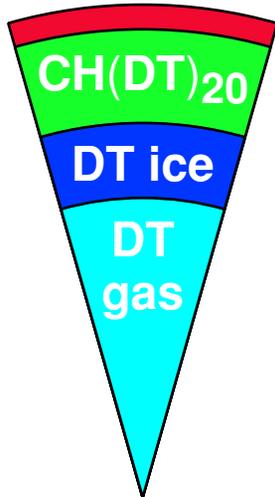
**$\xi/\xi_c$  of  $\alpha = 5$  and  $\alpha = 6$  can be reduced  
by increasing the in-flight aspect ratio**

$\alpha = 5$

<b>IFAR</b>	<b>46</b>	<b>56</b>
<b><math>A_{\text{bubble}}/\text{Th}</math> (%)</b>	<b>17</b>	<b>40</b>
<b><math>\xi/\xi_c</math></b>	<b>62</b>	<b>47</b>
<b>1-D gain</b>	<b>27</b>	<b>27</b>
<b>2-D gain</b>	<b>12</b>	<b>18</b>

# Three high-gain “wetted foam” designs have been considered

**Polyimide**



**Design 1:**

$$G = 124$$

$$V_{\text{imp}} = 3 \times 10^7 \text{ cm/s}$$

$$\rho R_m = 1.7 \text{ g/cm}^2$$

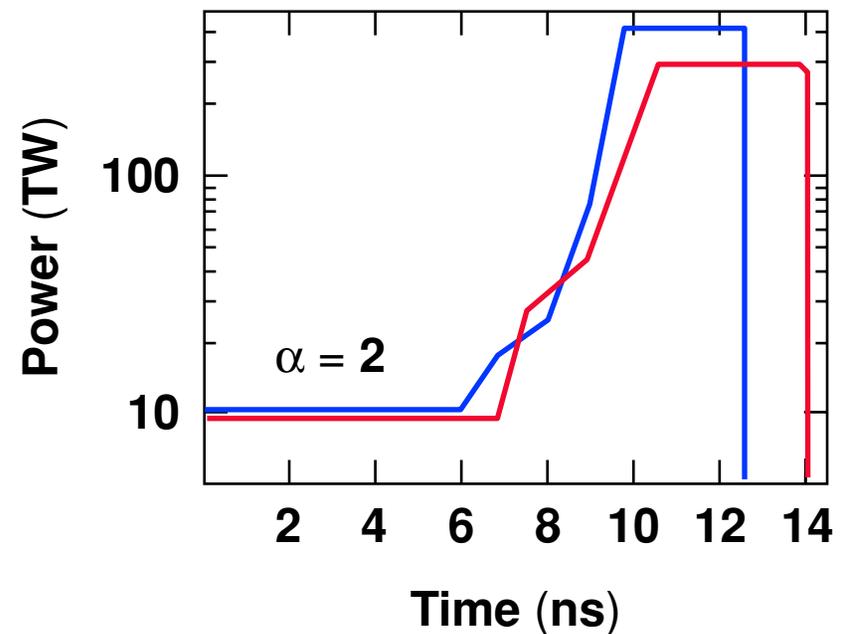
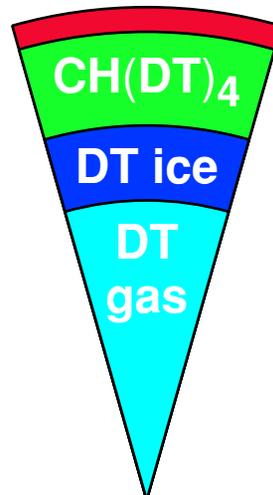
**Design 2:**

$$G = 81$$

$$V_{\text{imp}} = 3.9 \times 10^7 \text{ cm/s}$$

$$\rho R_m = 1.4 \text{ g/cm}^2$$

**Polyimide**



# Target stability and gain are calculated by using a developed model

Design	1	2
$A_{\text{bubble}}/\text{Th}$ (%)	22	52
$\xi/\xi_c$ (%)	200	72
1-D gain	124	81
2-D gain	0	55

- Assumptions:**
- (1) imprint is the same for “all-DT” designs
  - (2) perfect power balance
  - (3) 1- $\mu\text{m}$  DT-ice roughness and 80-nm outer surface finish

## Summary/Conclusion

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