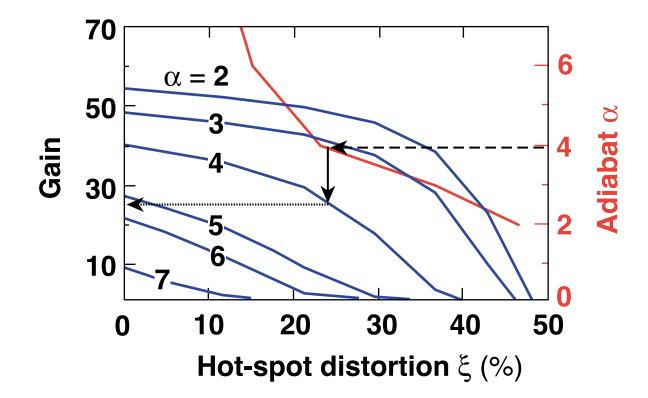
Hydrodynamically stability and gain of moderateto high-gain direct drive target designs for the NIF



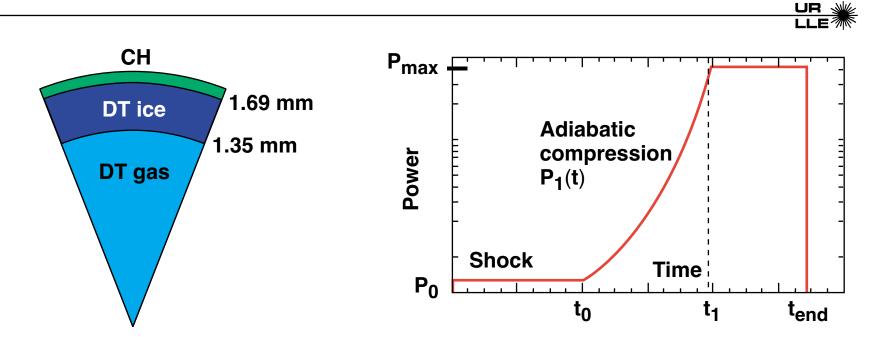
V. N. Goncharov, S. Skupsky, R. Betti, J. Marozas, P. McKenty and R. Town University of Rochester Laboratory for Laser Energetics 31st Annual Anomalous Absorption Conference Sedona, AZ 3–8 June 2001

Summary

A model has been developed to optimize NIF DD target designs

- A model has been developed to optimize target gain.
 - The model uses results of a stability postprocessor to calculate shell integrity during the acceleration phase and mode spectrum at shell stagnation.
 - Target gain is calculated by using the obtained mode spectrum and results of 1-D simulations with reduced implosion velocities.
- The model was applied to predict stability and gains for "all-DT" moderate-gain and high-gain foam target designs.
- The results of the model suggest that the maximum gain for the "all-DT" targets can be achieved for $\alpha = 3$ to $\alpha = 4$ designs.

"All-DT" DD NIF targets driven on adiabat up to 7 were considered



Pressure
$$P_{kidder} = \frac{P_0}{\left[1 - (t/\tau)^2\right]^{5/2}}$$

Power $P_1 = \frac{P_0}{\left[1 - (t/\tau)^2\right]^4}$

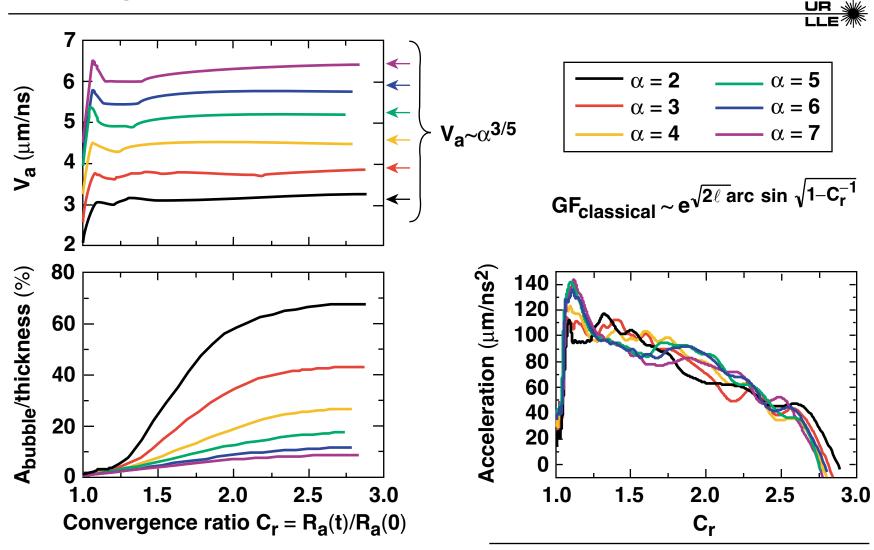
Parameter	Determined by
P ₀	Shell adiabat
P _{max}	Damage threshold, shell stability
t ₀	Timing of compression wave and first shock
t ₁	Target gain
t _{end}	Laser energy

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"All-DT" DD NIF targets driven on adiabat up to 7 were considered (continued)

500 V_{imp} ρ**R_{peak</mark>**} 200 (g/cm^2) $(\times 10^7 \text{ cm/s})$ Gain α 100 Power (TW) 2 1.5 4.17 55 50 3 1.3 4.27 48 **α** = **7** 4 1.2 4.34 41 5 1.1 4.42 29 20 6 1.0 4.42 22 10 7 0.9 4.45 9 2 2 6 8 10 12 0 4 Time (ns)

A stability postprocessor¹ was applied to study perturbation evolution of imploding targets during the acceleration, coasting, and deceleration phases

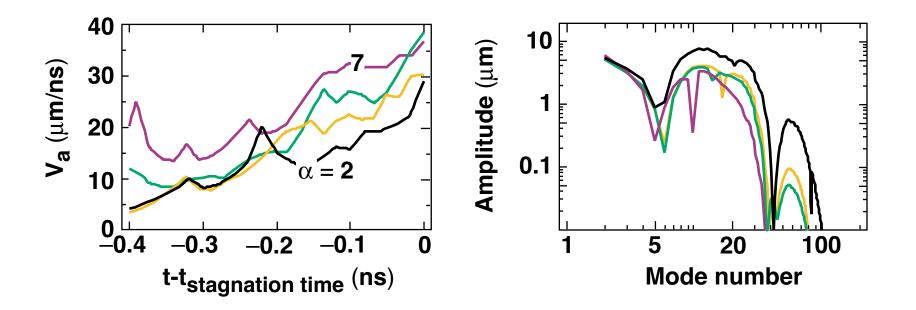


¹V. Goncharov et al., Phys. Plasmas <u>7</u>, 5118 (2000).

The postprocessor was used to calculate mode spectrum at stagnation

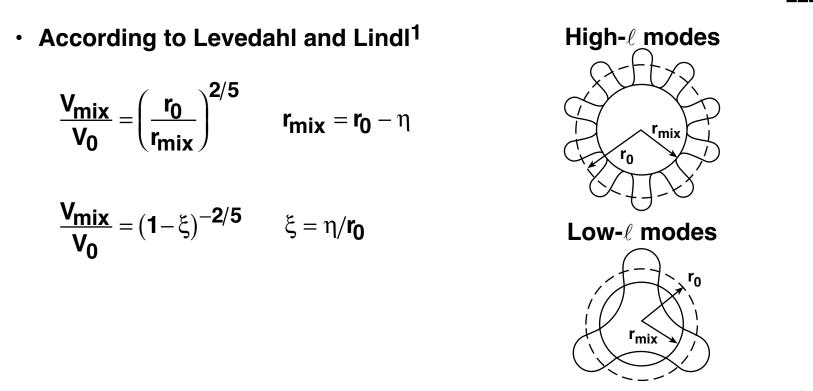
- V_a in decel phase is calculated by using theory of R. Betti¹ et al.
 - $V_a \propto \frac{\left(T_{hs}\right)^{5/2}}{R_{hs} \rho_{shell}}$

 Mode spectrum at the back surface of cold fuel at stagnation (1 THz SSD, 1 μm DT ice roughness, 800Å outer surface finish)



¹V. Lobatchev and R. Betti, Phys. Rev. Lett., <u>85</u>, 4522 (2000).

The mode spectrum at stagnation is related to the gain reduction

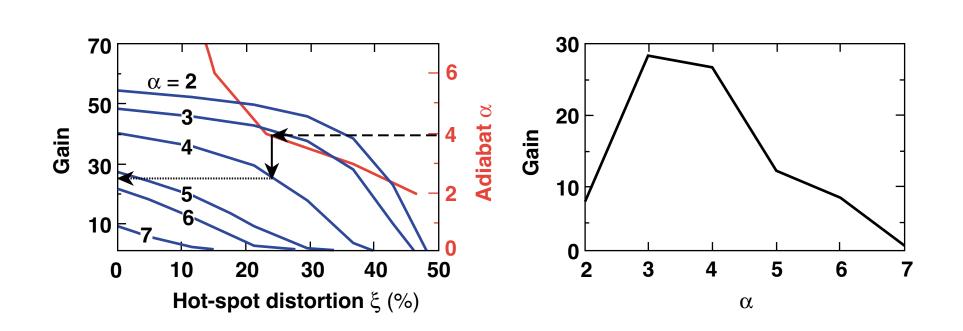


Perturbation is equivalent to a reduction in 1-D implosion velocity:²

$$\xi = 1 - \left(\frac{V_0 - \Delta V}{V_0}\right)^{5/2}$$

¹W. Levedahl and J. Lindl, Nuc. Fusion <u>37</u>, 165 (1997). ²Roy Kishony, Ph.D. thesis, 1999.

Gain is calculated by using the results of 1-D simulations with reduced V_{imp}



- Target gain is calculated assuming 1-THz, 2-D SSD;
 1-μm ice–DT gas roughness; and 800-Å outer surface finish.
- The imprint spectrum is assumed to be the same for different α 's.

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