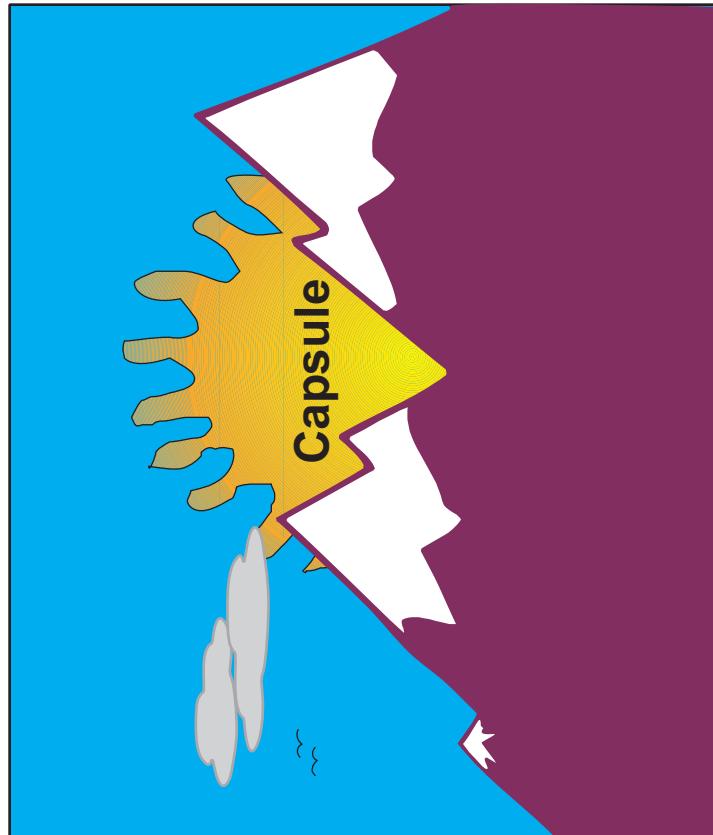
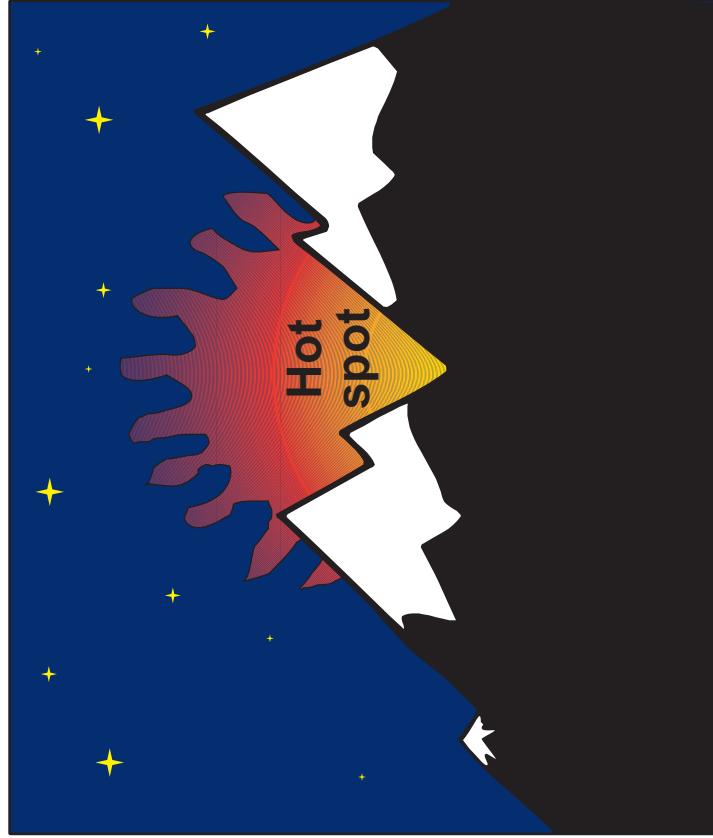


Hydrodynamic Instabilities from the Beginning to the End



Implosion at dawn



Implosion at sunset

R. Betti, V. Goncharov, J. Knauer,
V. Lobatchev, and M. Umanski
University of Rochester
Laboratory for Laser Energetics

30th Annual Anomalous
Absorption Conference
Ocean City, MD
21–26 May 2000

Acceleration or deceleration, this is the question!



Shock transit and acceleration phase

- RM and RT growth
- RT and RM seeding:
 - outer-surface nonuniformities
 - inner-surface nonuniformities (feedout)
 - laser imprintings

Reflected shock transit and deceleration phase

- RM and RT growth
- RT and RM seeding:
 - outer-surface nonuniformities (feedout)
 - inner-surface nonuniformities (feedthrough)

Why all this theory is so incredibly useful!

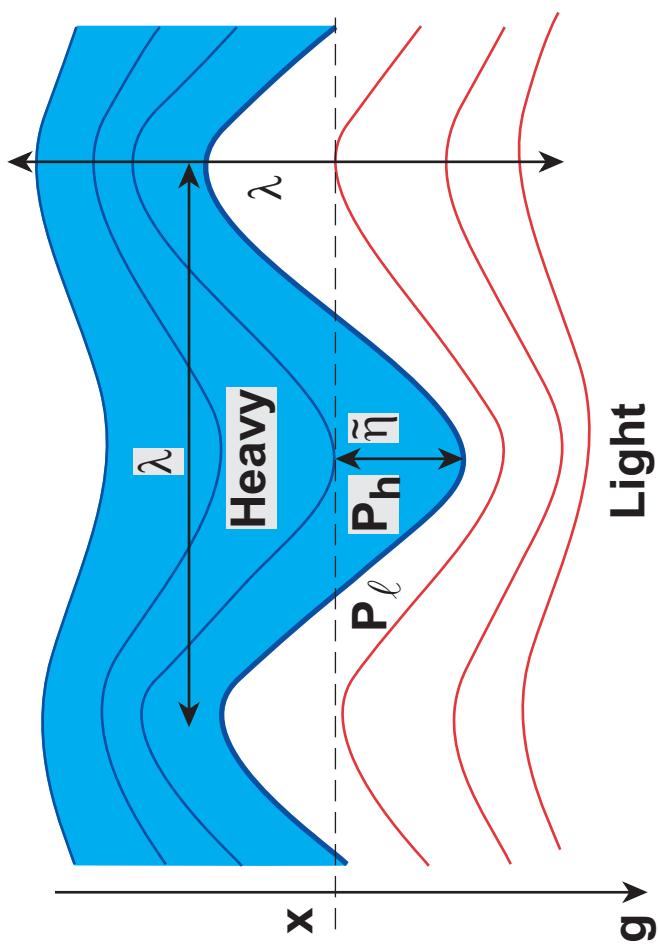
Why all this theory is so incredibly useful!



- Keeps theorists employed.
- Improves physical understanding.
- Provides feedback to numerical simulations.
- Leads to the development of a fast, reliable, and accurate postprocessor of 1-D codes. Target design can be carried out using such a postprocessor and subsequently refined using 2-D simulations.

Acceleration Phase

The classical RT is just Newton's law at work: $F = ma!$



$$F = S(P_h - P_\ell) = ma = \rho_h \lambda \ddot{\tilde{\eta}}$$

$$\frac{dP_0}{dx} = \rho_0 g = \left\{ \begin{array}{ll} \rho_h g & \text{heavy} \\ \rho_\ell g & \text{light} \end{array} \right\}$$

$$P_\ell = P_0 + \left[\frac{dP_0}{dx} \right]_\ell \tilde{\eta} = P_0 + \rho_\ell g \tilde{\eta}$$

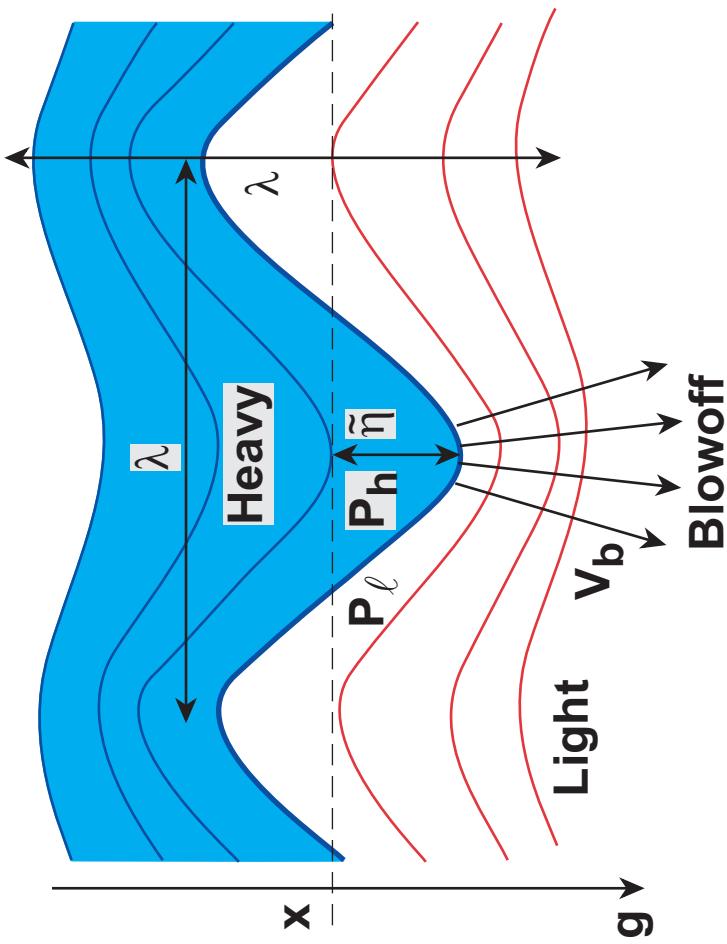
$$P_h = P_0 + \left[\frac{dP_0}{dx} \right]_h \tilde{\eta} = P_0 + \rho_h g \tilde{\eta}$$

$$F = ma \rightarrow S \rho_h g \tilde{\eta} = \rho_h \lambda S \ddot{\tilde{\eta}} \rightarrow \ddot{\tilde{\eta}} = k g \tilde{\eta} \rightarrow \tilde{\eta} \sim e^{\gamma t} \rightarrow \gamma = \sqrt{k g}$$

$k \sim 1/\lambda$

Acceleration Phase

The ABLATIVE RT is just Newton's law at work again but with a restoring force: the dynamic pressure



- Newton's law

$$S[P_h - (P_\ell + \rho_\ell V_b^2)] = \rho_h \lambda \tilde{\eta}$$

- Energy balance

$$\rho V_b = q_{heat} \rightarrow \tilde{V}_b = V_b \tilde{\eta} / \lambda$$

- Perturbed dynamic pressure

$$\underbrace{\rho_\ell V_b}_{\text{Ablation rate}} \tilde{V}_b = \dot{m} V_b \tilde{\eta} / \lambda$$

Restoring force

Stabilizing

$$\bullet \text{Growth rate: } S(\rho_h g - k \dot{m} V_b) \tilde{\eta} = \rho_h \lambda \ddot{S} \tilde{\eta} \rightarrow \tilde{\eta} \sim e^{\gamma t} \rightarrow \gamma = \sqrt{k g - k^2 \frac{\dot{m}}{\rho_h} V_b}$$

Acceleration Phase

The “old”-fashioned ablative stabilization is still there
but it is not very effective



- A more accurate calculation yields additional stabilization:

$$\gamma = \sqrt{Akg - k^2 \frac{\dot{m}}{\rho_h} V_b + 4k^2 V_a^2 - 2kV_a}$$

“Old”-fashioned
Dynamic pressure

$$\bullet \text{ Atwood number: } A = \frac{\rho_{\text{heavy}} - \rho_{\text{light}}(\lambda)}{\rho_{\text{heavy}} + \rho_{\text{light}}(\lambda)} \approx 1$$

- The cutoff wave number depends only on the dynamic pressure:

$$kg = k^2 \frac{\dot{m}}{A\rho_h} V_b \rightarrow k_{\text{cutoff}} = \frac{\rho_h g}{\dot{m} V_b A}$$

The ABLATIVE RM is NOT an instability but just damped oscillations

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Impulsive acceleration

- Newton's law:

$$[\rho_h \mathbf{g}(t) - k \dot{m} \mathbf{V}_b] \tilde{\eta} = \rho_h \lambda \ddot{\tilde{\eta}}$$

- For $t = 0^+$:

$$\omega = \sqrt{k^2 \frac{\dot{m}}{\rho_h} V_b}$$

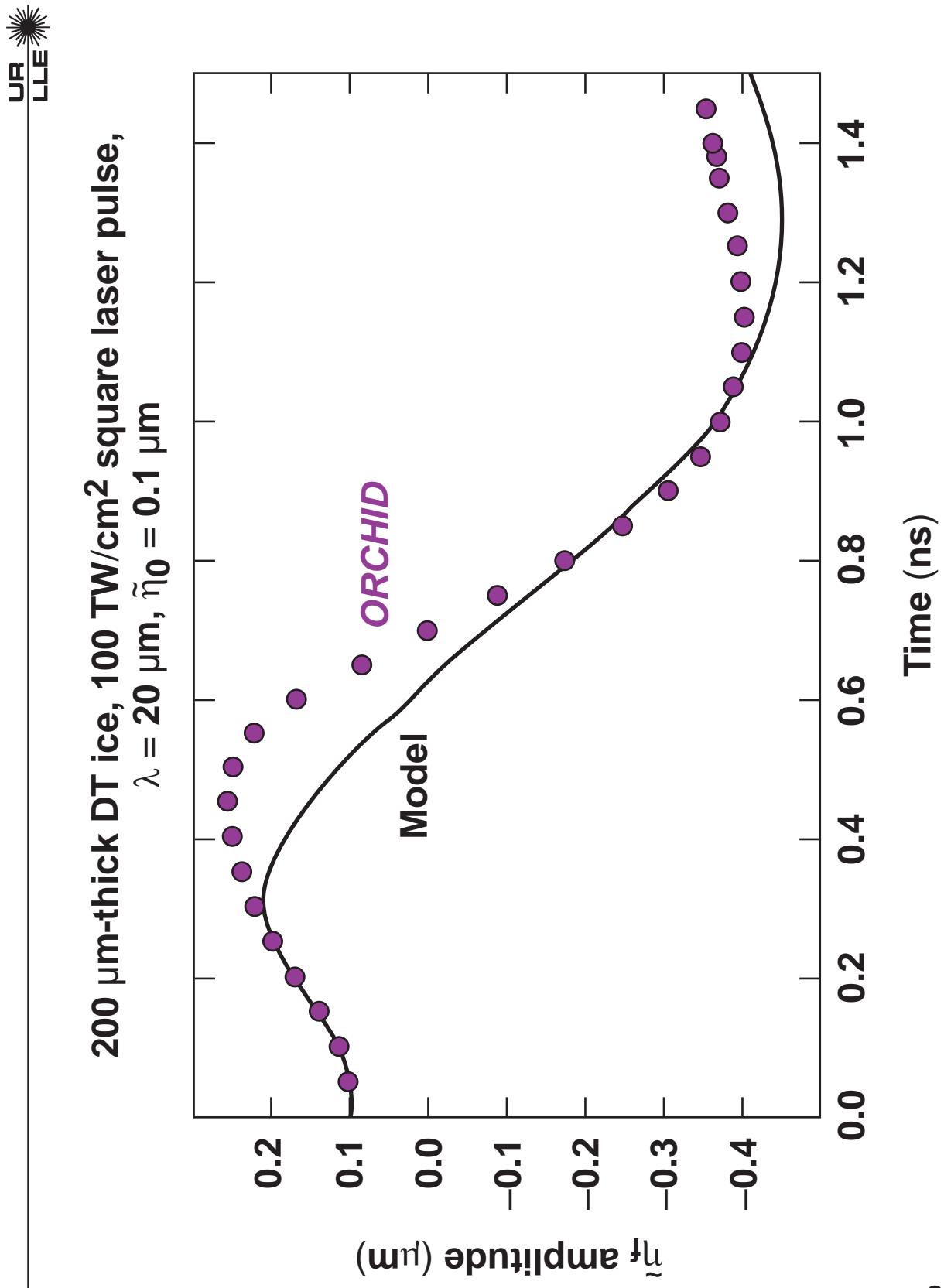
- For $t > 0^+$, $\mathbf{g}(t) = 0 \rightarrow \tilde{\eta}(t) = \tilde{\eta}(0)^+ \cos(\omega t) + \frac{\dot{\tilde{\eta}}(0)^+}{\omega} \sin(\omega t)$

- If the ablative convection of vorticity is included then the RM is damped:

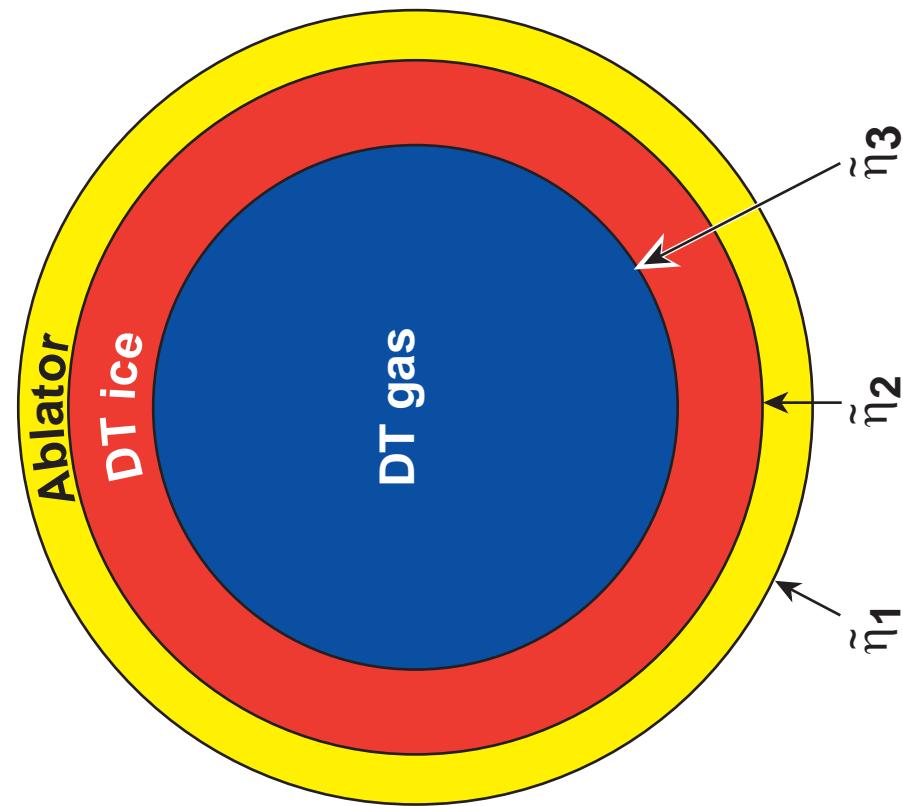
$$\tilde{\eta}(t) = \left\{ \tilde{\eta}(0) \cos(\omega t) + \frac{\dot{\tilde{\eta}}(0) + 2kV_a \tilde{\eta}(0)}{\omega} \sin(\omega t) \right\} e^{-2kV_a t}$$

A. Velikovich et al., Phys. Plasmas 5, 1491 (1998); V. Goncharov, Phys. Rev. Lett. 82, 2091 (1999).

Prediction of the sharp boundary model agrees with the numerical results



The RT postprocessor is based on the sharp boundary model



- $\tilde{\eta}$ represents the surface distortion.
- The model is based on a set of coupled second-order differential equations:

$$\sum_{j=1}^N a_j(t) \ddot{\tilde{\eta}}_j + b_j(t) \dot{\tilde{\eta}}_j + c_j(t) \tilde{\eta}_j = 0$$

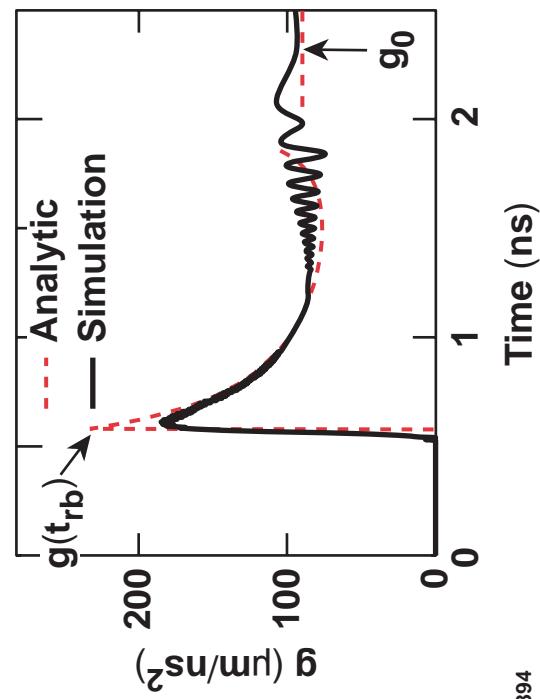
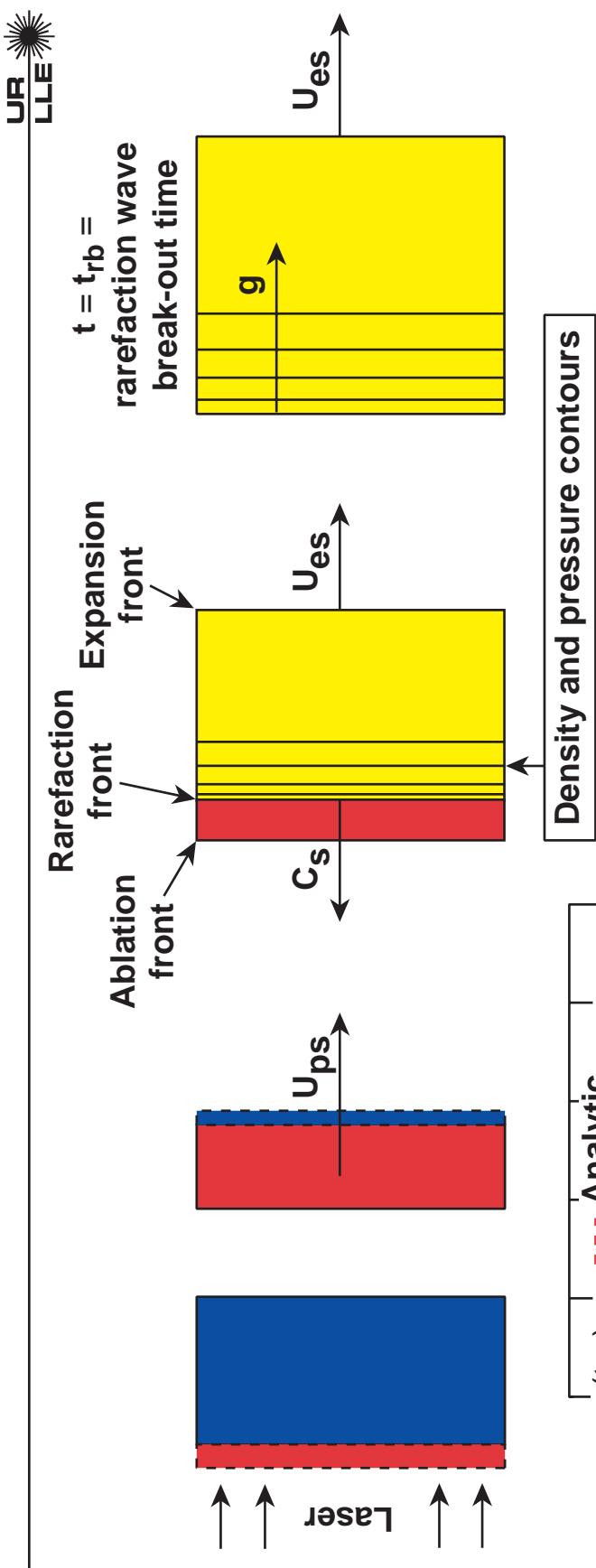
- Physics included
 - thermal conduction
 - ablative flow
 - compressibility
- The solution of the model requires the initial conditions: $\tilde{\eta}_j(0)$, $\dot{\tilde{\eta}}_j(0)$
- All the details of sharp boundary model can be found in V. Goncharov's PhD thesis (UR 1998).

RT seeding by rear-surface nonuniformities



Feedout

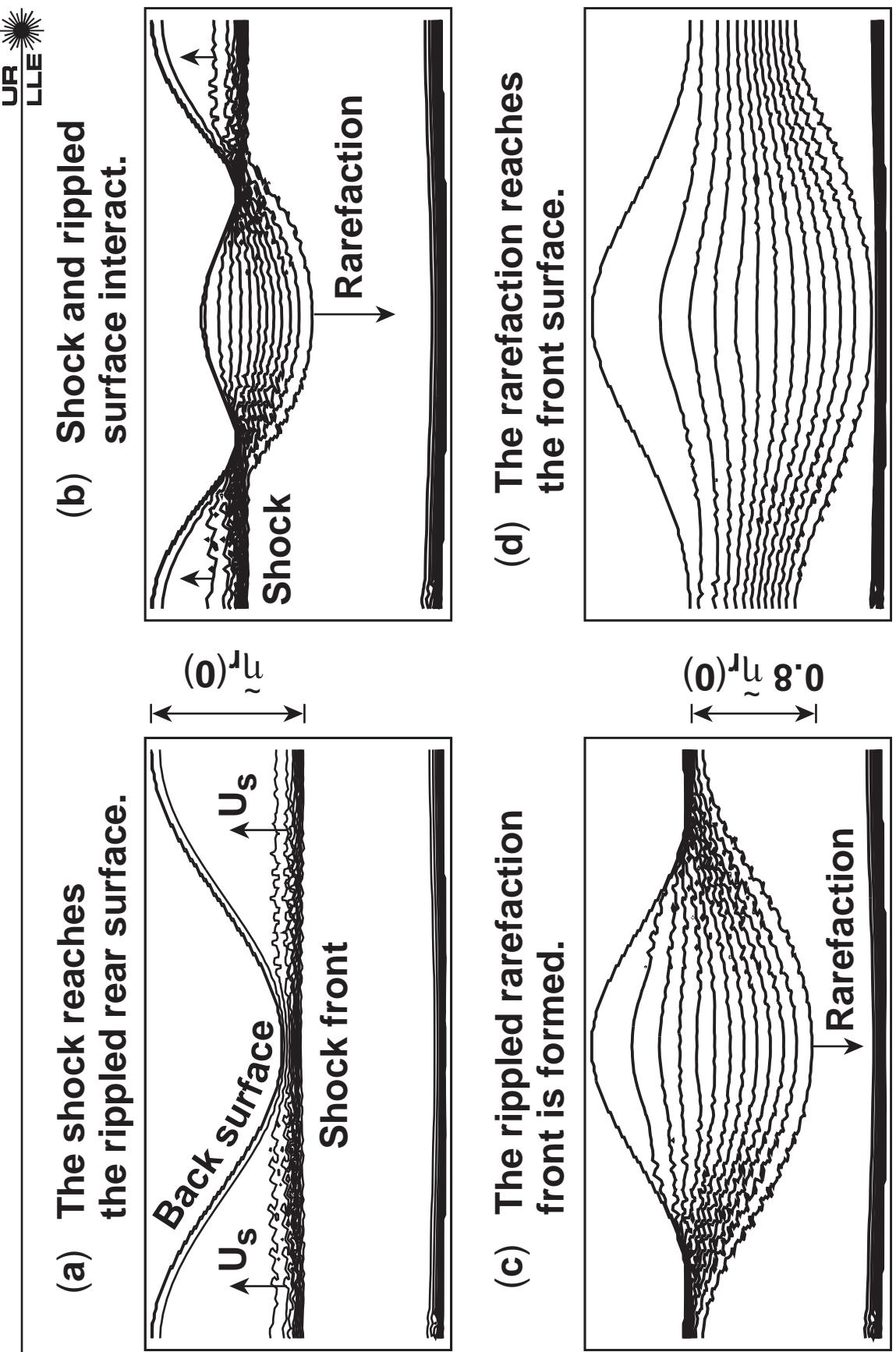
After the shock breaks out, a rarefaction wave propagates toward the ablation front while the rear surface expands; after the rarefaction wave breaks out, the ablation front accelerates



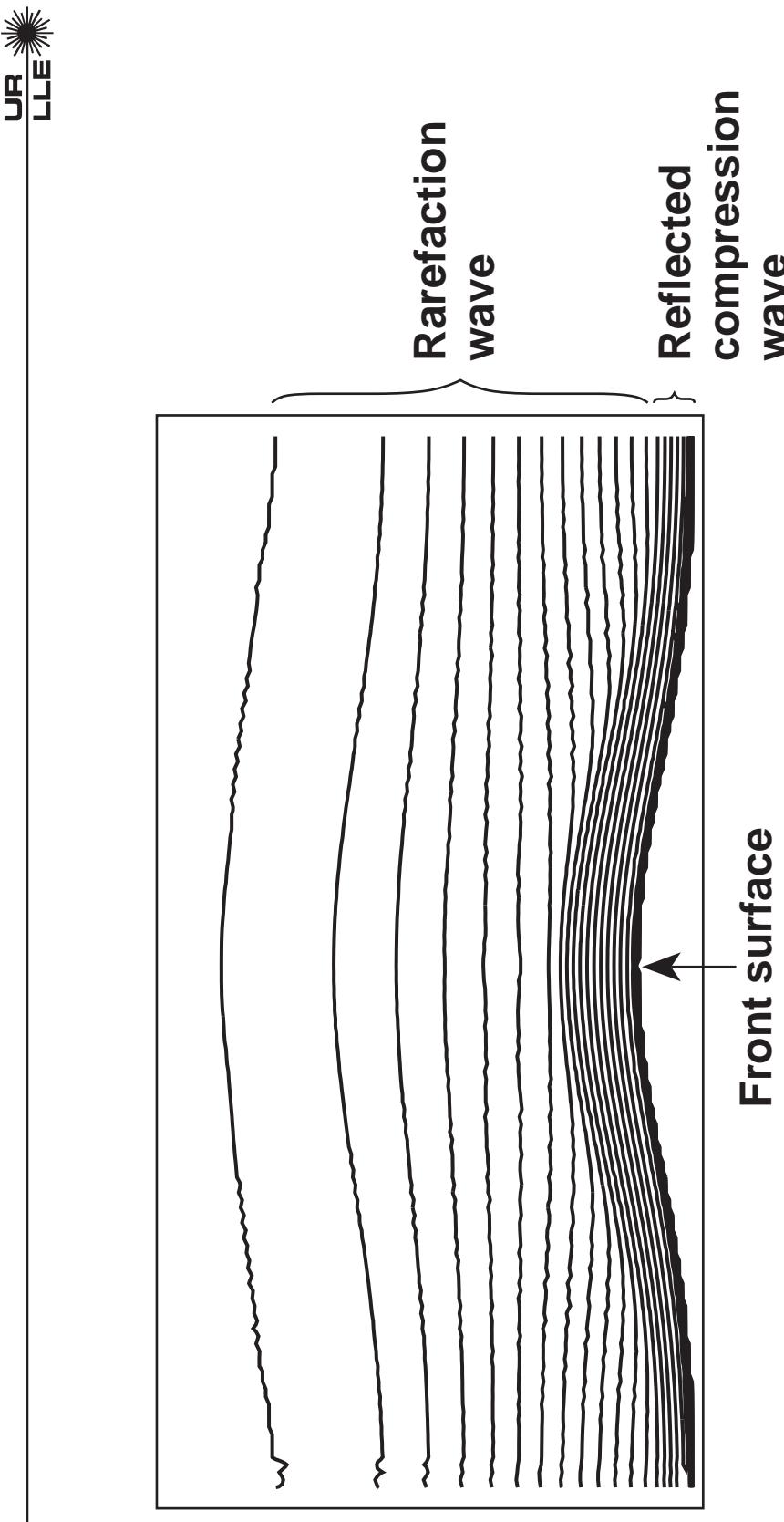
$$g(t_{rb}) = 2.5 g_0$$

$$g_0 \equiv g(\text{steady state}) = \frac{P_a}{\rho d}$$

Two-dimensional behavior in the presence of a rippled rear surface



After the rarefaction breakout the ripple on the front surface is seeded and begins to grow

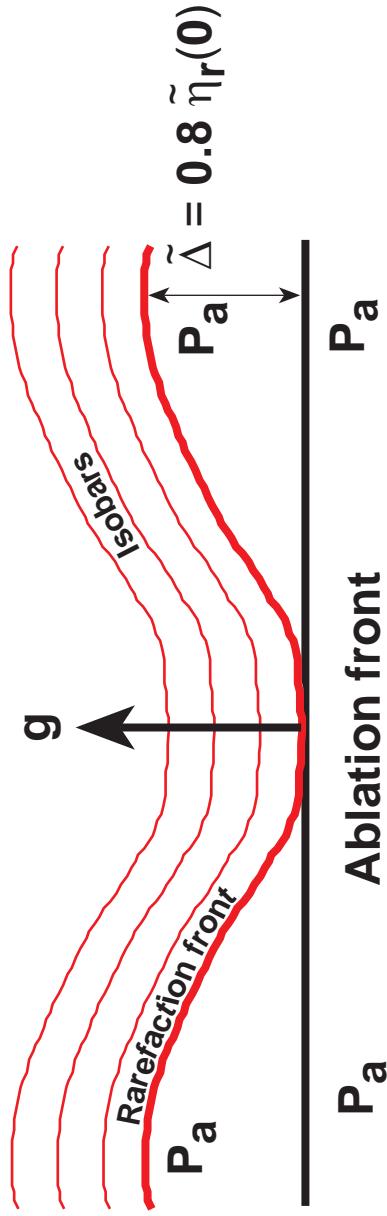


0.1 ns after rarefaction breakout

When the rippled rarefaction wave reaches the ablation front, it imprints a velocity perturbation and the ablation front develops a ripple that starts growing linearly in time



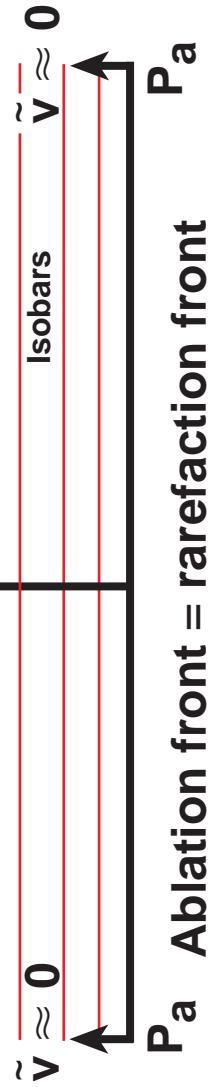
$$t = t_{rb} = \text{rarefaction-wave break-out time}$$



$$g(t_{rb}) = \frac{5}{2} \frac{P_a}{\rho d}$$

$$\tilde{v} = g \Delta t$$

$$t = t_{rb} + \Delta t$$



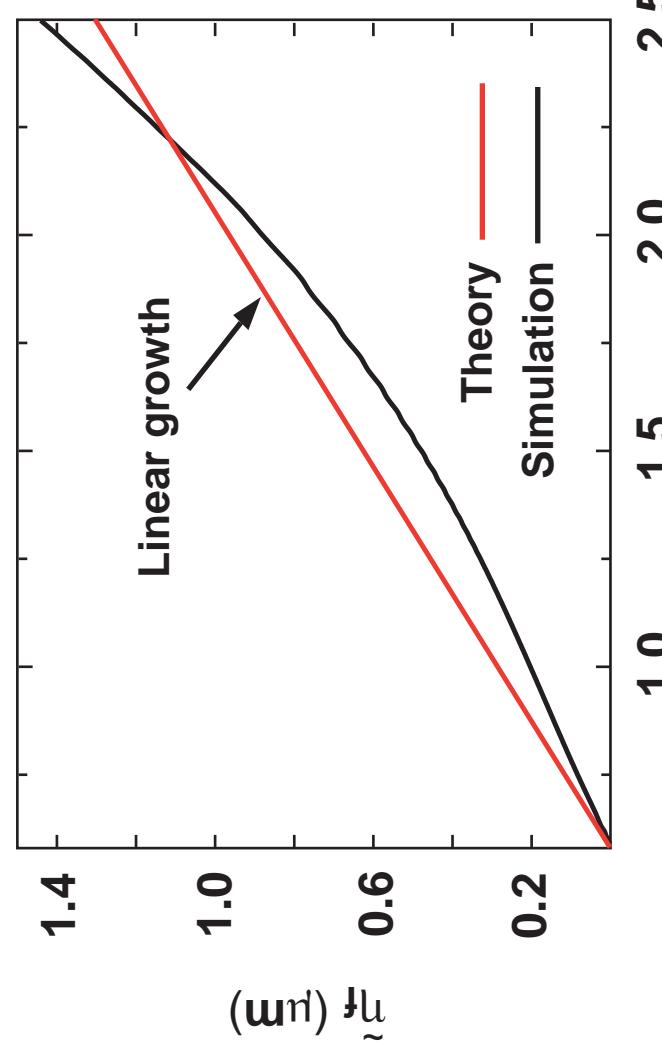
$$\tilde{v} = \frac{6}{5} C_s \frac{\tilde{\eta}_r(0)}{d_{comp}}$$

→ This theory is valid only for $k d_{comp} < 1$.

The predicted ablation-front-surface ripple-amplitude evolution soon after the rarefaction wave breaks out does NOT accurately reproduce the simulation results



Ablation-front-surface distortion versus time

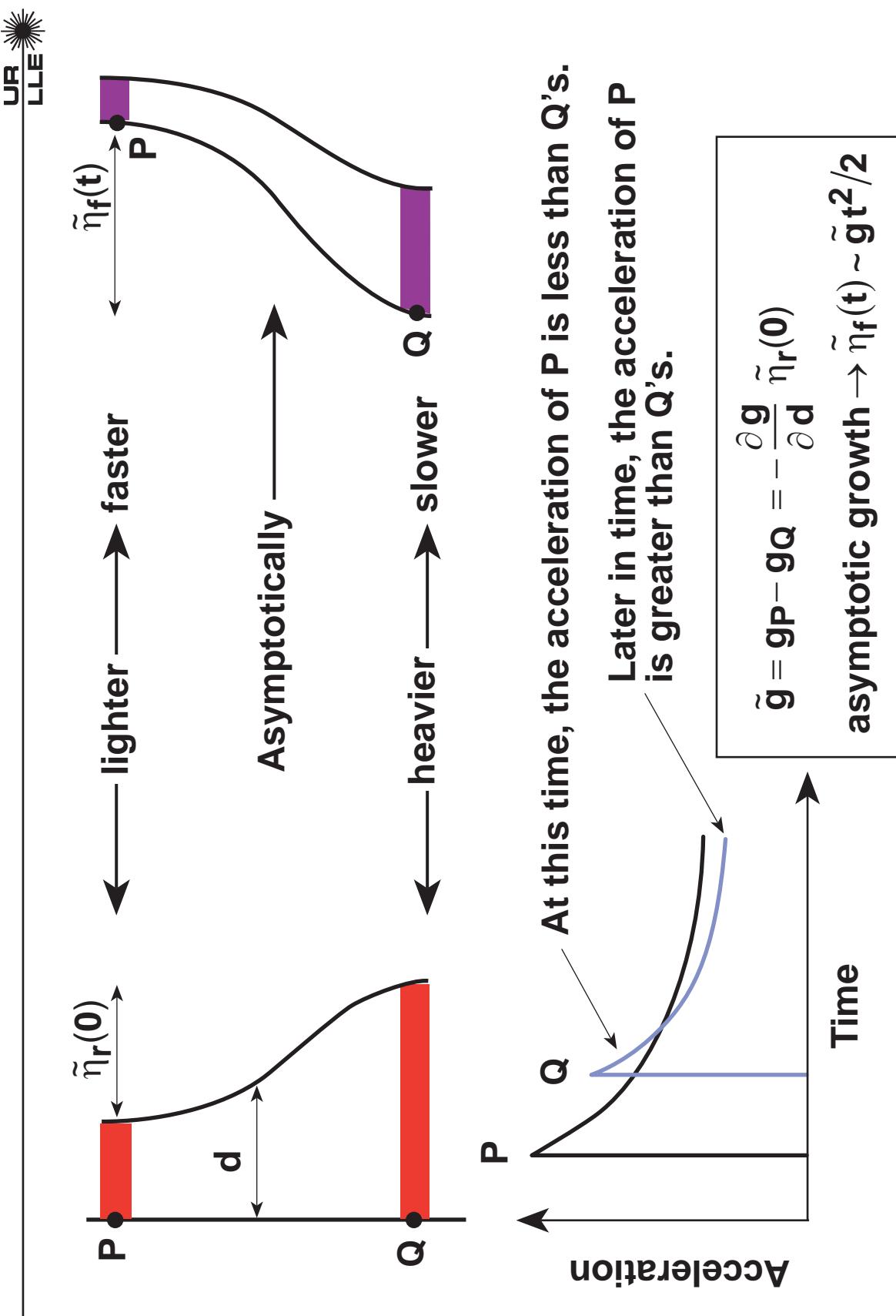


The theory predicts a linear growth right after the rarefaction wave breaks out.

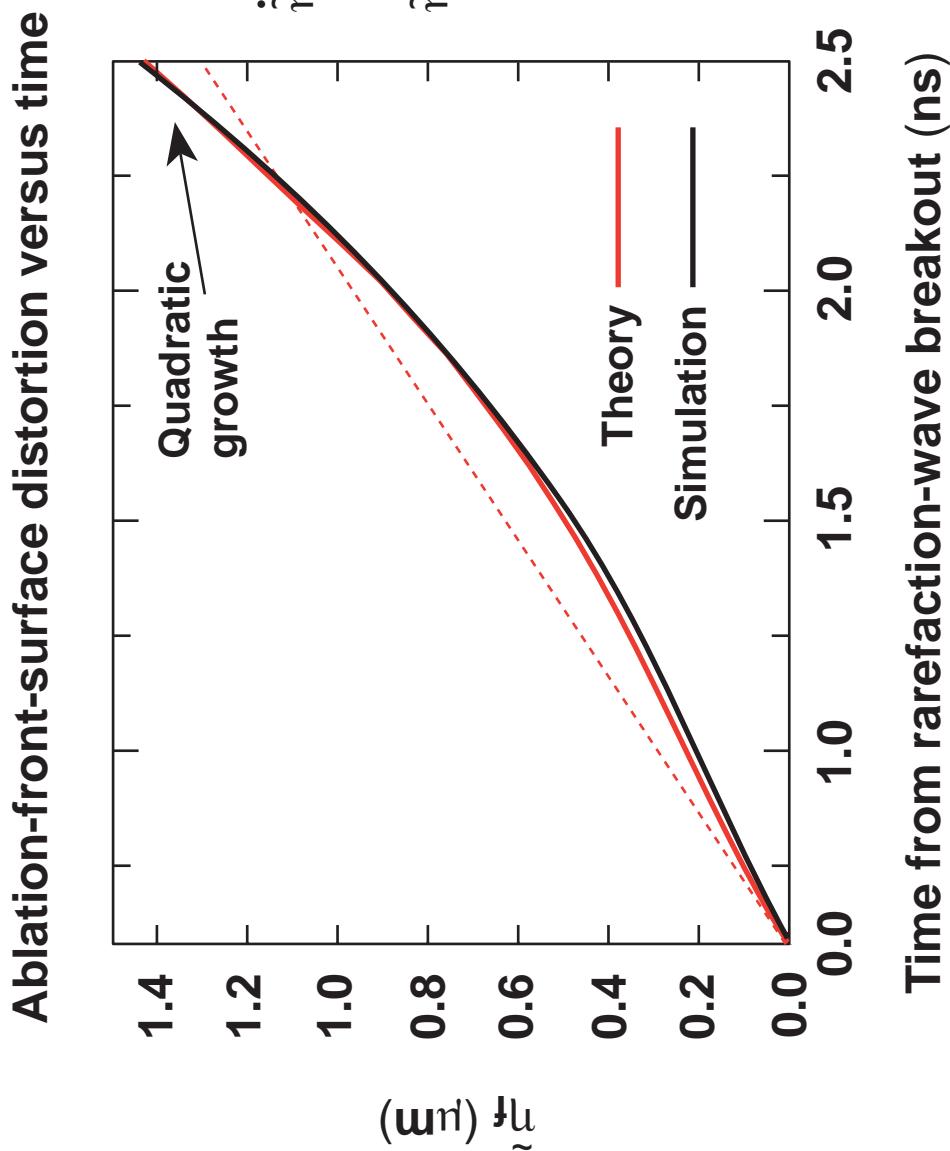
$$\begin{aligned}\tilde{\eta}_f &= \tilde{v} t \\ \tilde{\eta}_r(0) &= 0.1 \mu\text{m} \\ \tilde{v} &\approx 0.7 \mu\text{m/ns} \\ k d_{\text{comp}} &= 0.033 \\ P &= 20 \text{ Mbar} \\ d(\text{CH}) &= 20 \mu\text{m} \\ d_{\text{comp}} &\approx d/4\end{aligned}$$

Time from rarefaction wave breakout (ns)

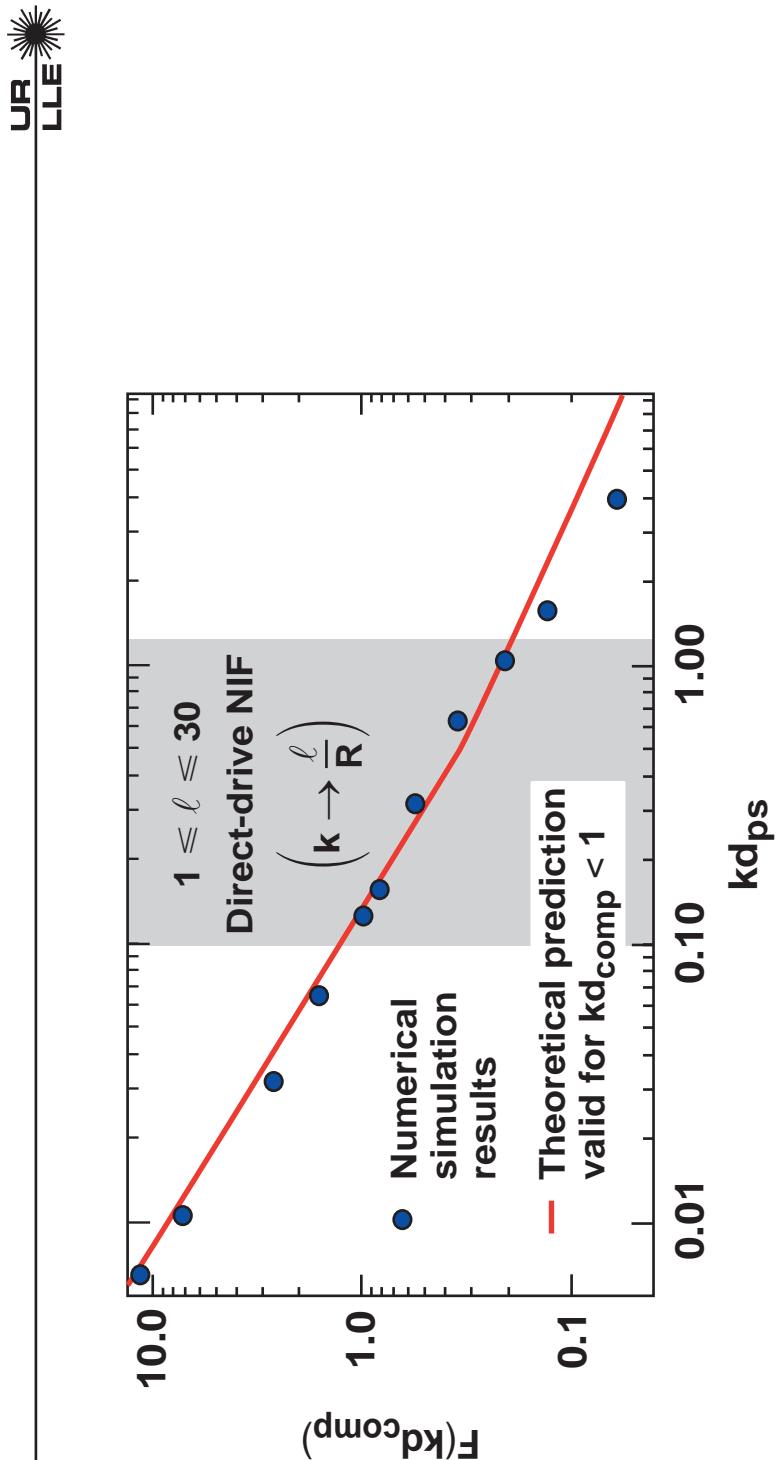
In addition to the linear growth, an approximately quadratic growth occurs because of the perturbed acceleration



The predicted ablation-front-surface ripple-amplitude evolution is in good agreement with the simulation results when the perturbed acceleration is included



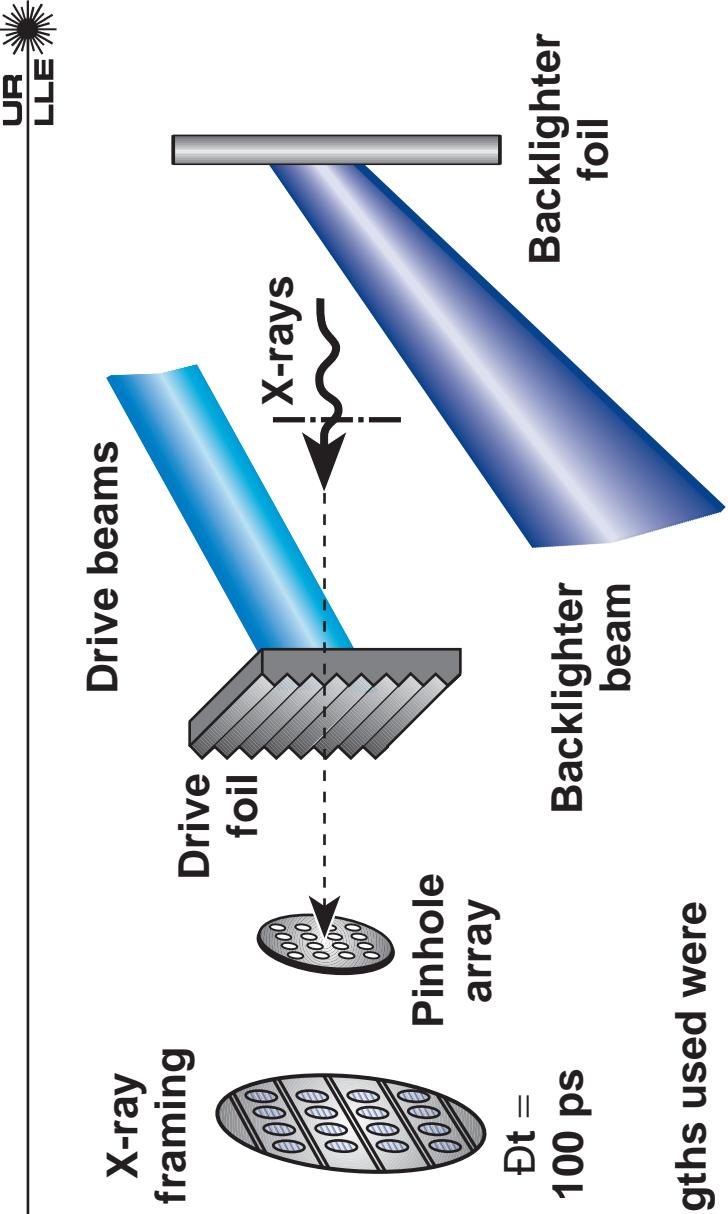
Both theory and simulations indicate that the transfer function for NIF is well below unity for $\ell > 30$



$$\eta_f(t) = \eta_r(0) F(k_{d\text{comp}}) \exp \left[\int_{t_{rb}}^t \sqrt{k g(t')} dt' \right]$$

- Transfer function from theory $F(k_{d\text{comp}}) = \frac{1}{16} \left[\frac{1}{k_{d\text{comp}}} + \frac{2.5}{\sqrt{k_{d\text{comp}}}} \right]$

Feed-out of the rear-surface perturbation was measured for three wavelengths



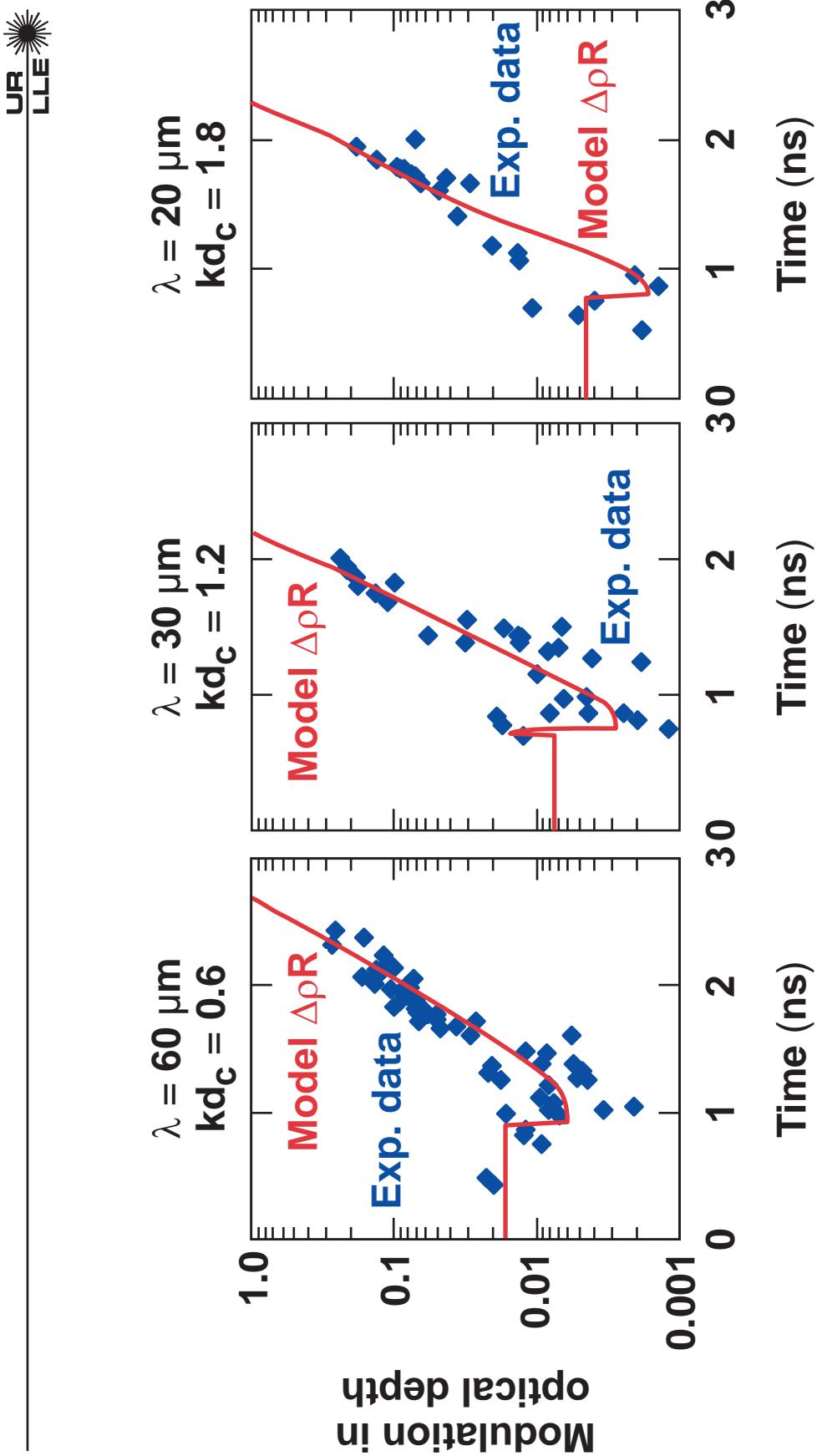
- Perturbation wavelengths used were

- 60 μm with a 0.5 μm amplitude
- 30 μm with a 0.5 μm amplitude
- 20 μm with a 0.5 μm amplitude

- Target foils were constructed from 20- μm -thick CH.

- Targets with 60- μm wavelength perturbations had a front-surface amplitude = 10% of rear-surface amplitude (0.05 μm).

Calculated ρR from sharp boundary model agrees with experimentally measured optical depth



- ρR is scaled by x-ray mfp and framing camera MTF.

RT seeding by laser-intensity nonuniformities

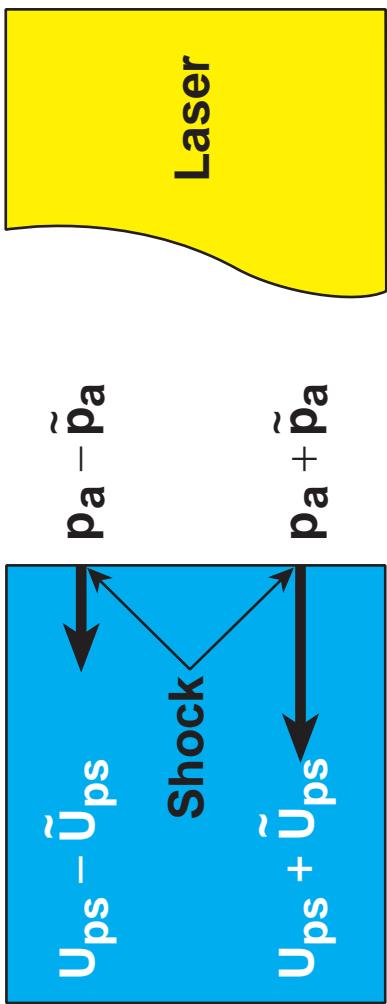


Laser Imprintings

Hydrodynamic flow is the main imprint mechanism: velocity perturbation

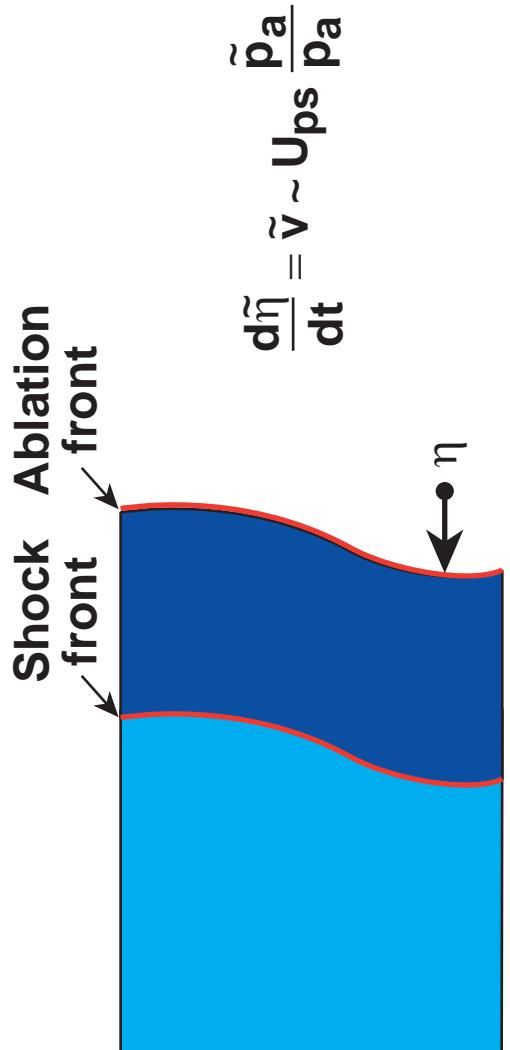
UR
LLE

- Post-shock speed depends on the ablation pressure $U_{ps} \sim \sqrt{p_a}$



$$\tilde{v} \sim \tilde{U}_{ps} \sim \frac{1}{2} U_{ps} \frac{\tilde{p}_a}{p_a}$$

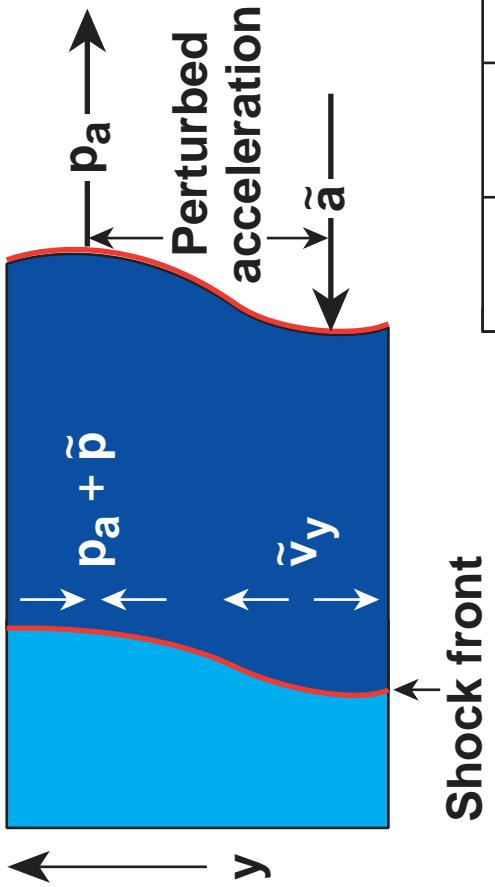
$$\tilde{\eta}_{vel} = \tilde{v} t \sim \frac{\tilde{p}_a}{p_a} c_{st}$$



Hydrodynamic flow is the main imprint mechanism: acceleration perturbation

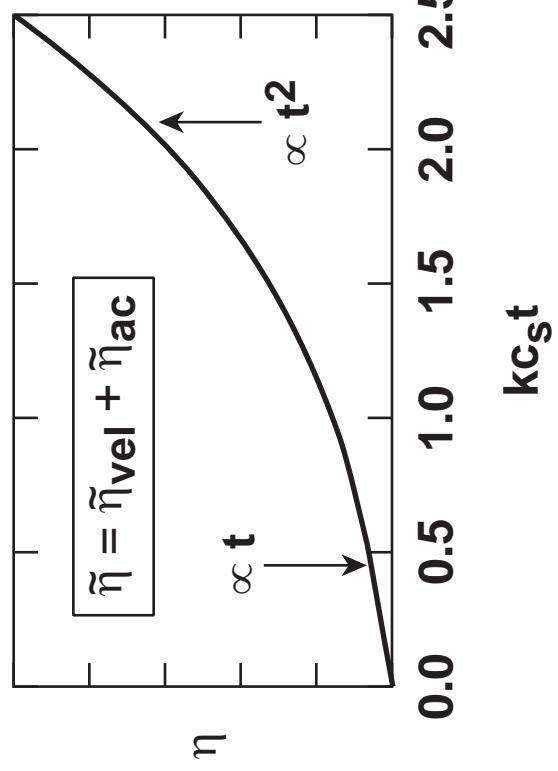
UR
LLE

- Rippled shock creates lateral mass flow.

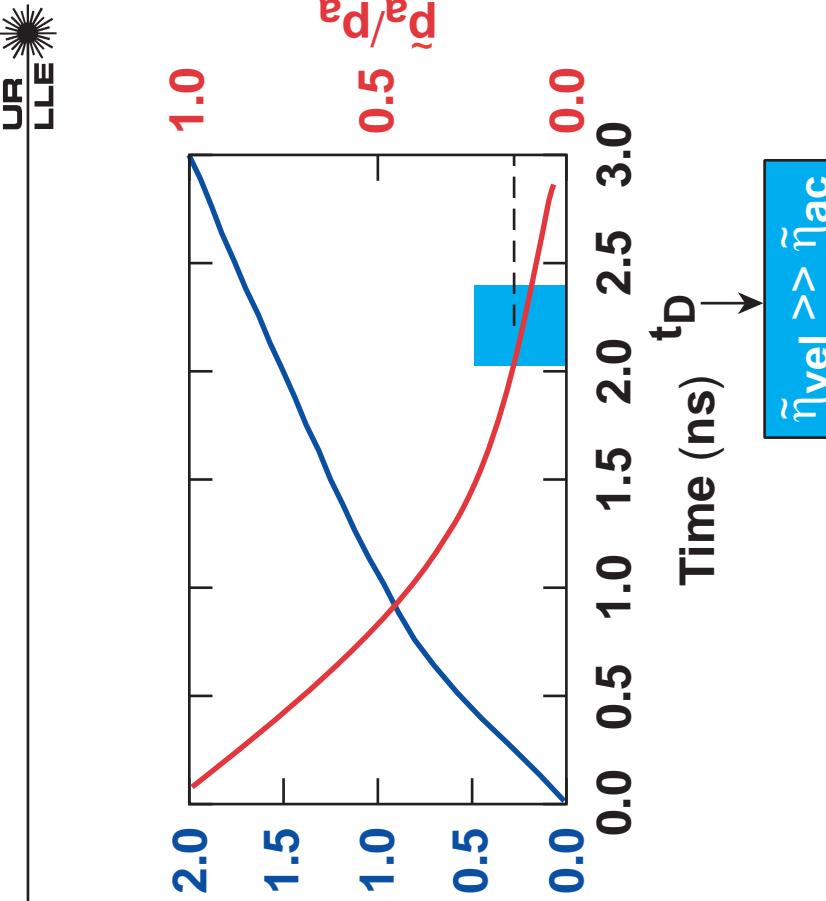
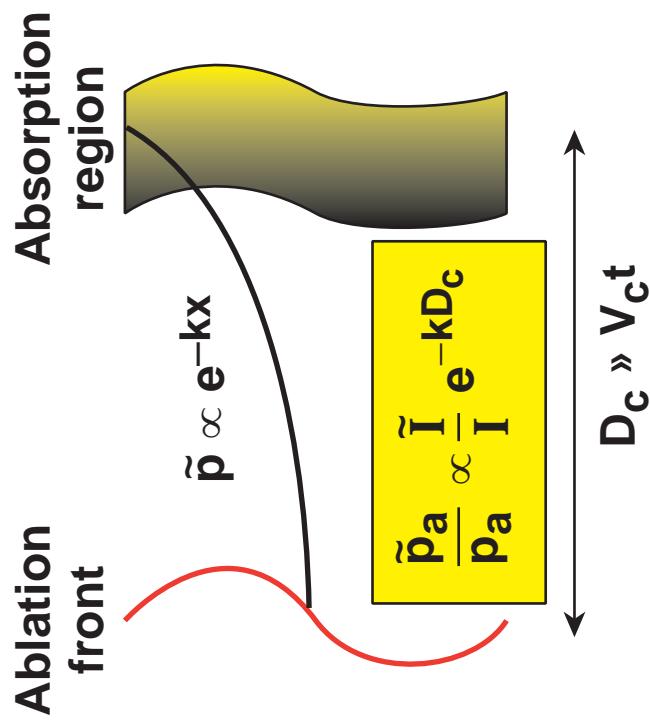


$$\eta_{ac} \propto k \frac{\tilde{p}_a}{p_a} c_s^2 t^2$$

$$\frac{d^2 \tilde{\eta}}{dt^2} = \tilde{a} \propto k \frac{\tilde{p}_a}{\rho}$$



Thermal smoothing¹ suppresses acceleration perturbations

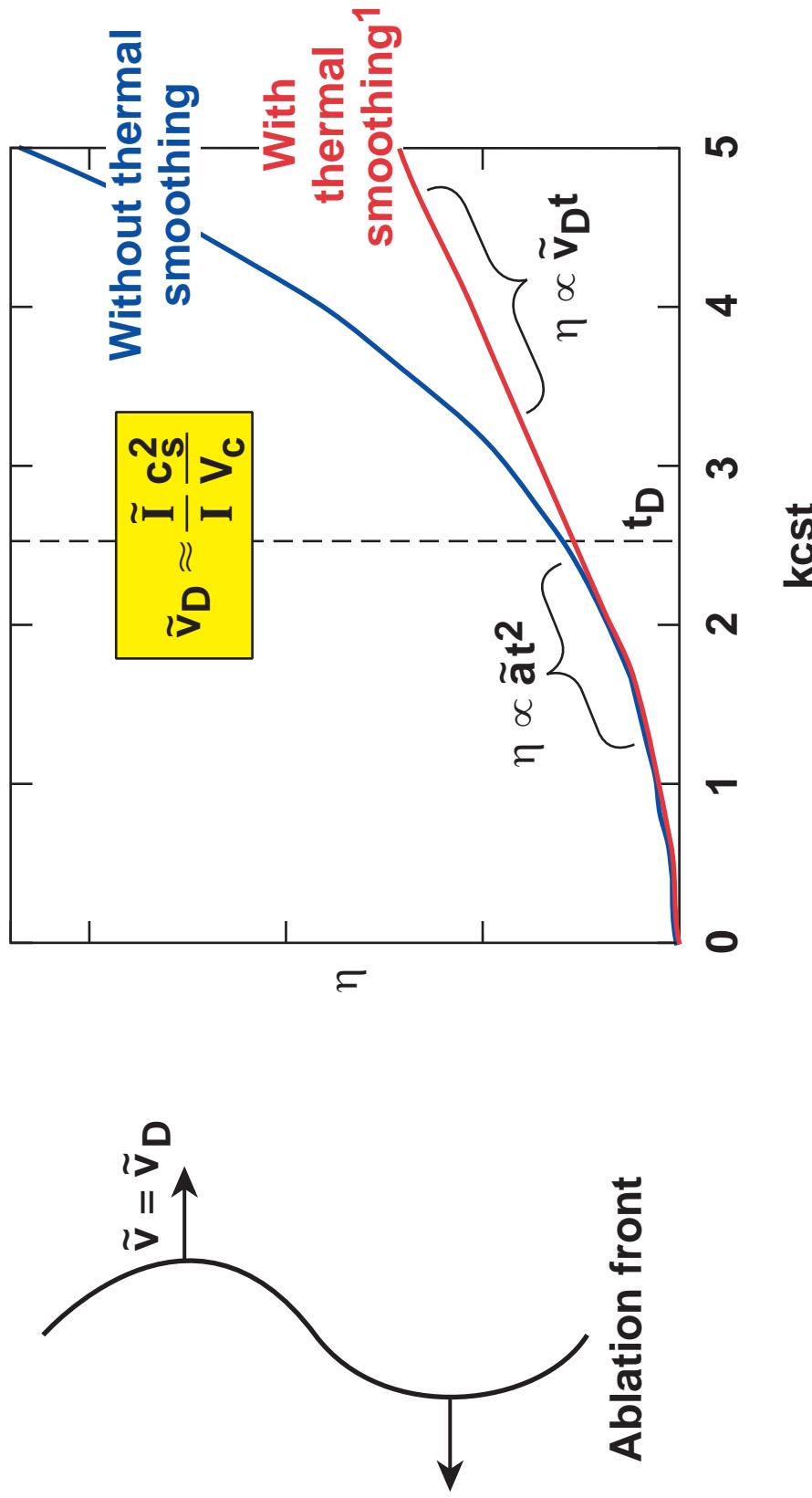


- Laser perturbations decouple from the ablation front when $kD_c \sim 1$.

Decoupling time $t_D \propto (kV_c)^{-1}$

Imprint growth is reduced by thermal smoothing

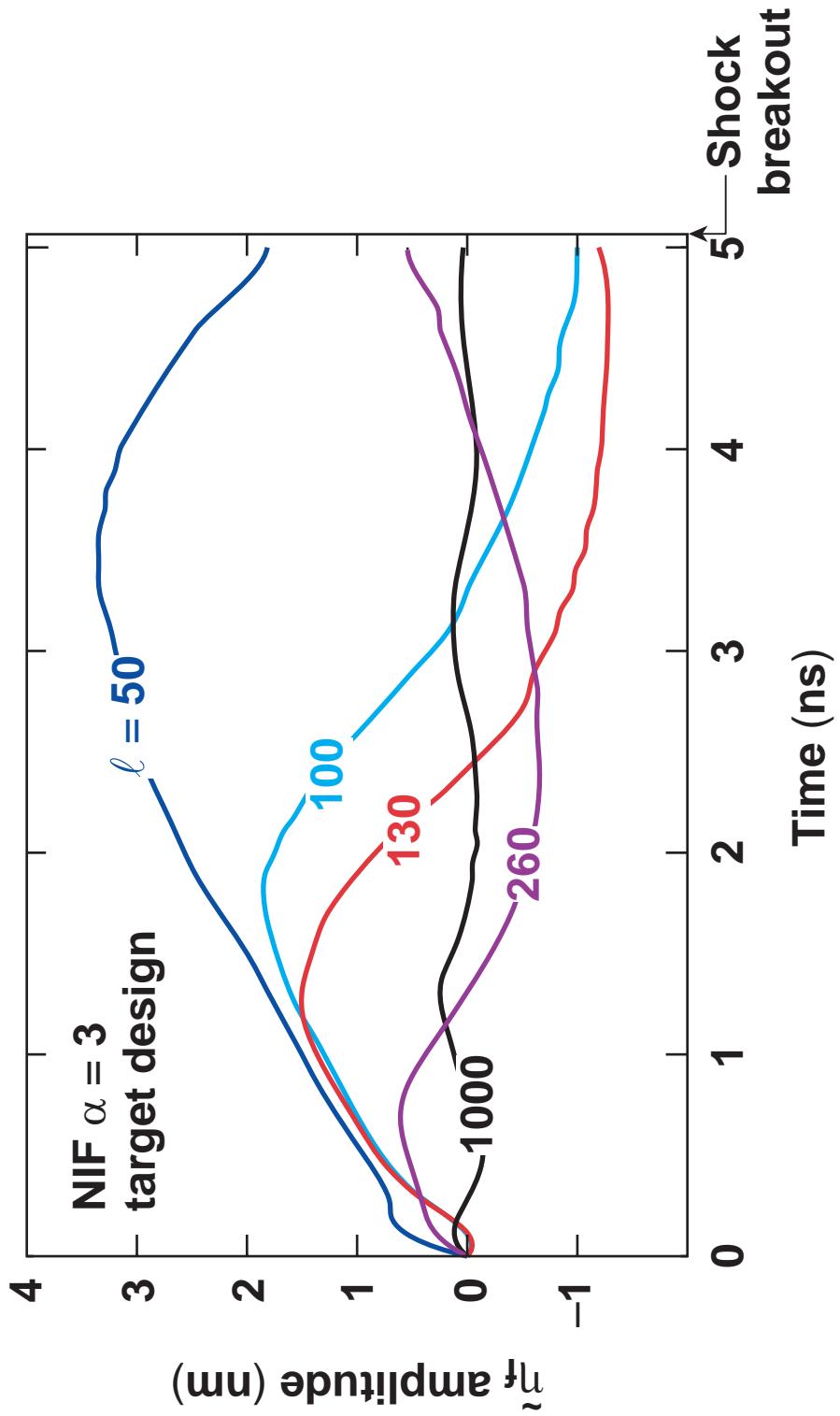
- After decoupling time $t > t_D$, $\tilde{a} = 0$.



The most damaging modes oscillate during the shock propagation

UR
LLE

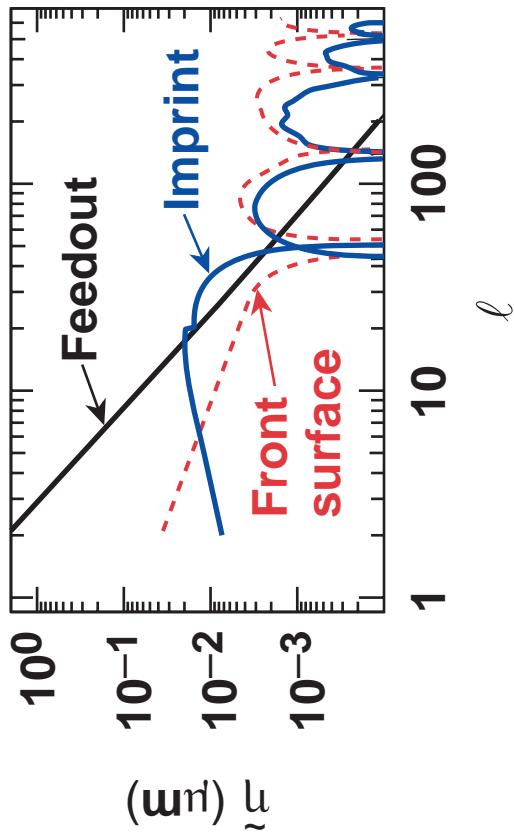
- Single-mode imprint *ORCHID* simulations



Mode spectrum at the beginning of the acceleration phase is used as an initial condition for RT postprocessor



- Initial conditions for RT model



- RT postprocessor

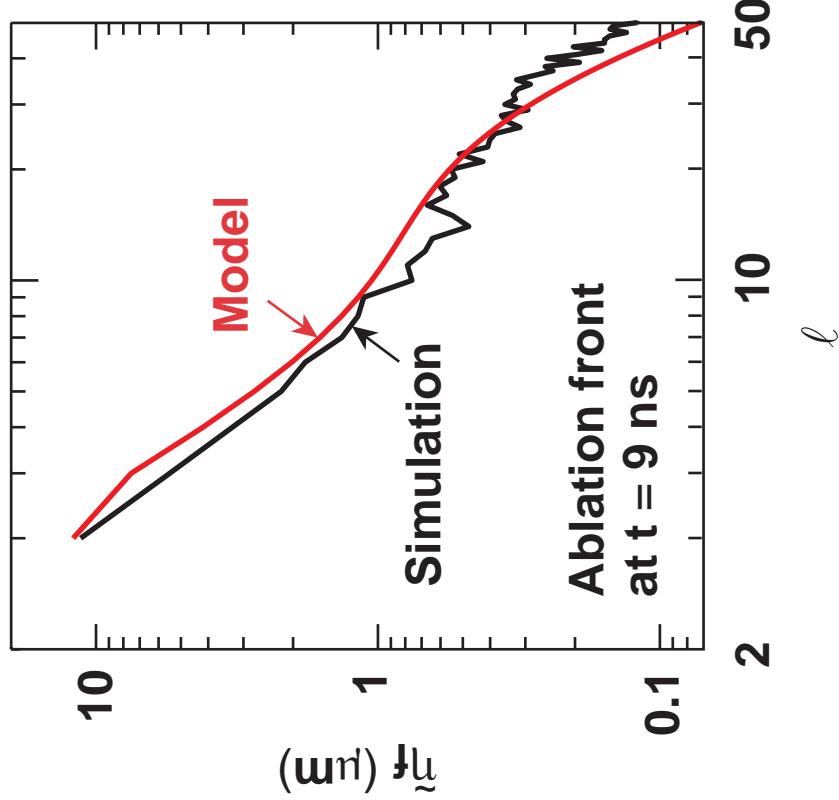
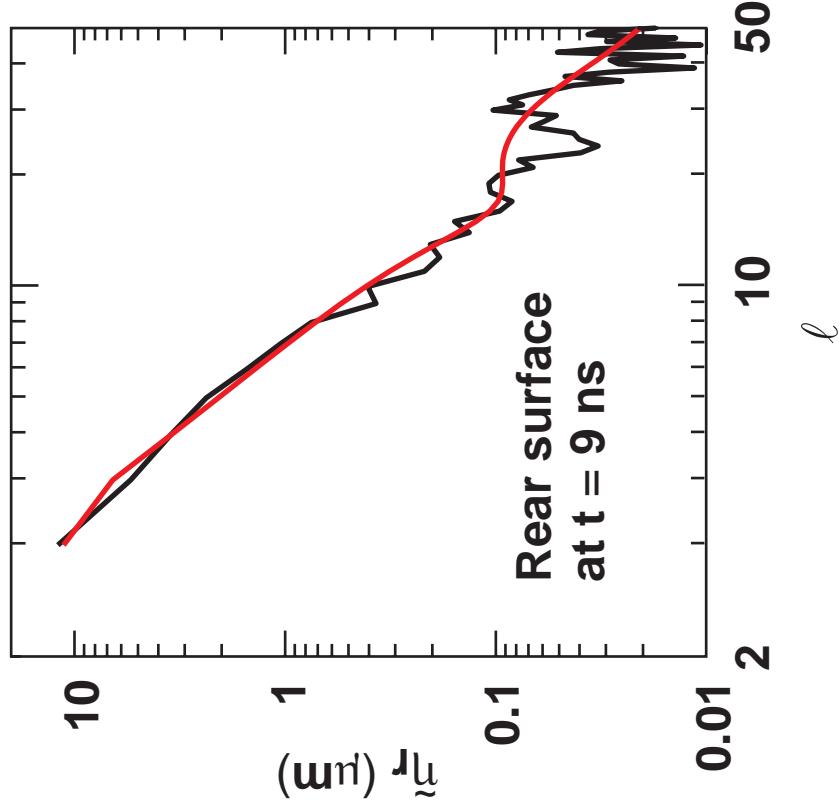
$$\left\{ \begin{array}{l} \sum_{j=0}^{2} \left[C_j \frac{d^j}{dt^j} \eta_f + D_j \frac{d^j}{dt^j} \eta_r \right] = 0 \\ \sum_{j=0}^{2} \left[E_j \frac{d^j}{dt^j} \eta_f + F_j \frac{d^j}{dt^j} \eta_r \right] = 0 \end{array} \right.$$

The diagram shows a circular domain with a jagged boundary. Inside the circle, there is a light blue shaded region. On the left side of the boundary, there is a dark blue arrow pointing outwards, labeled $\tilde{\eta}_f$. On the right side of the boundary, there is a dark blue arrow pointing inwards, labeled $\tilde{\eta}_r$.

The results of the model are in good agreement with multimode *ORCHID* simulations



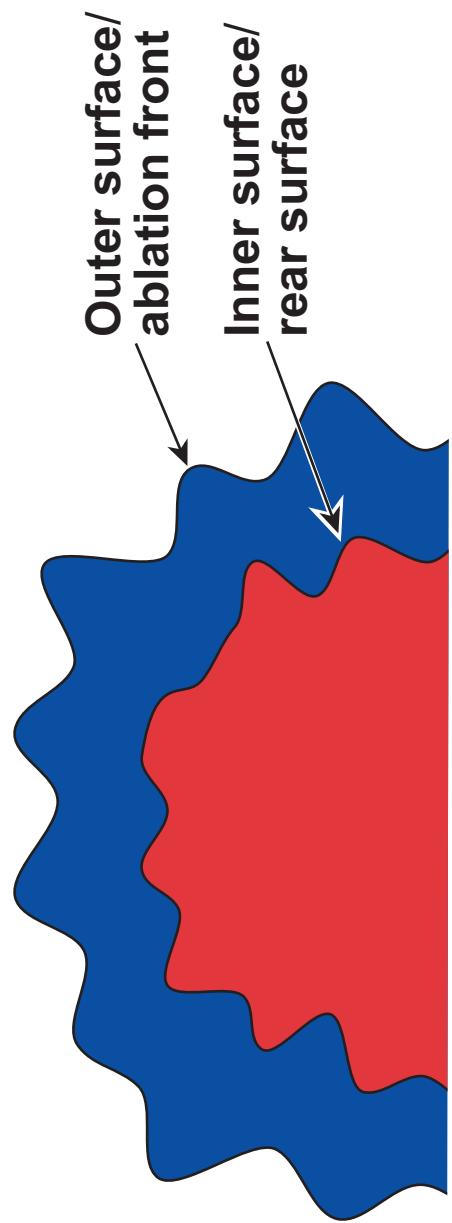
- Multimode perturbation at the beginning of the deceleration phase.



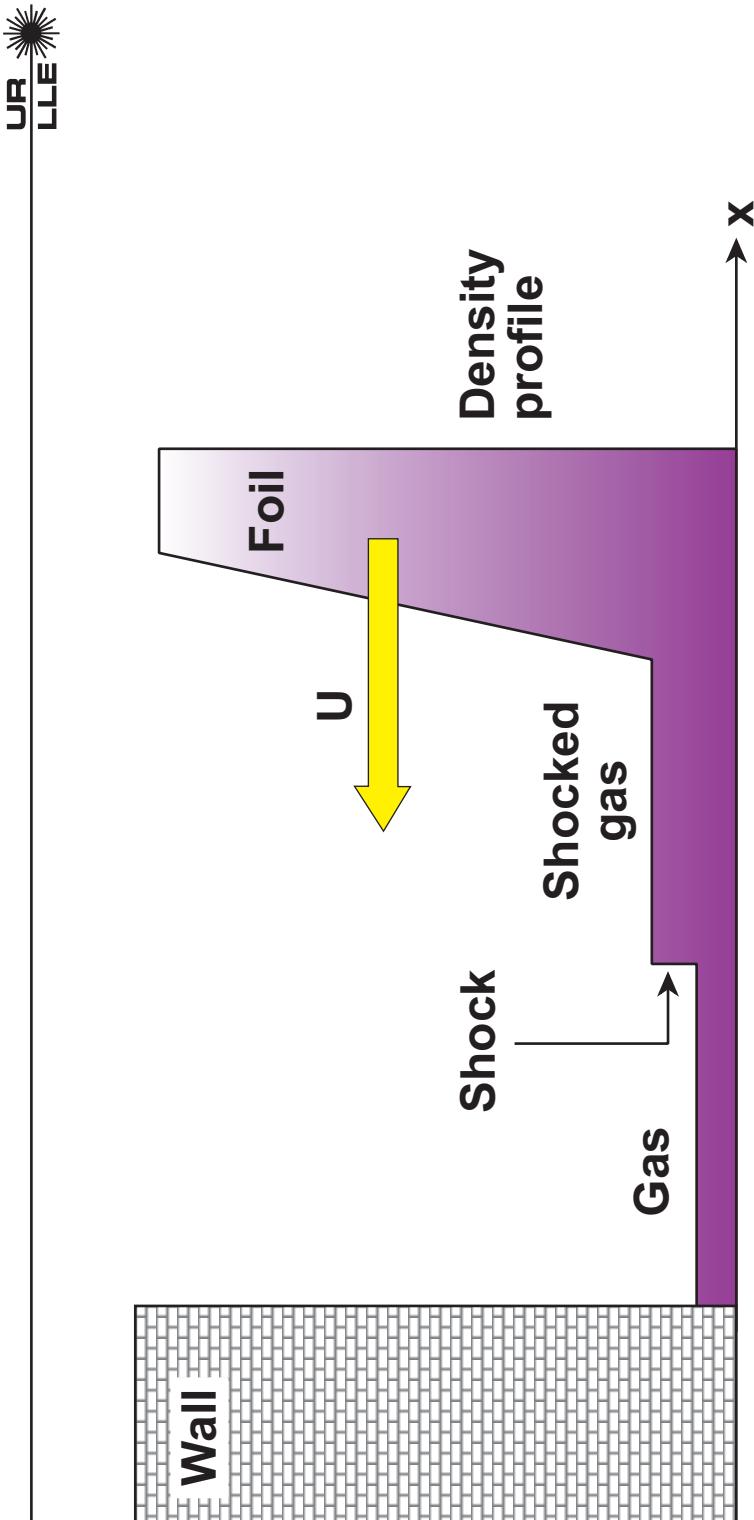
Deceleration phase instabilities

UR
LLE

- The simple model of a decelerating foil
- The seeds
- The growth rates
- Change of motation:
ablation front → outer surface
rear surface → inner surface.



The decelerating foil problem provides the basic understanding of the deceleration-phase instability

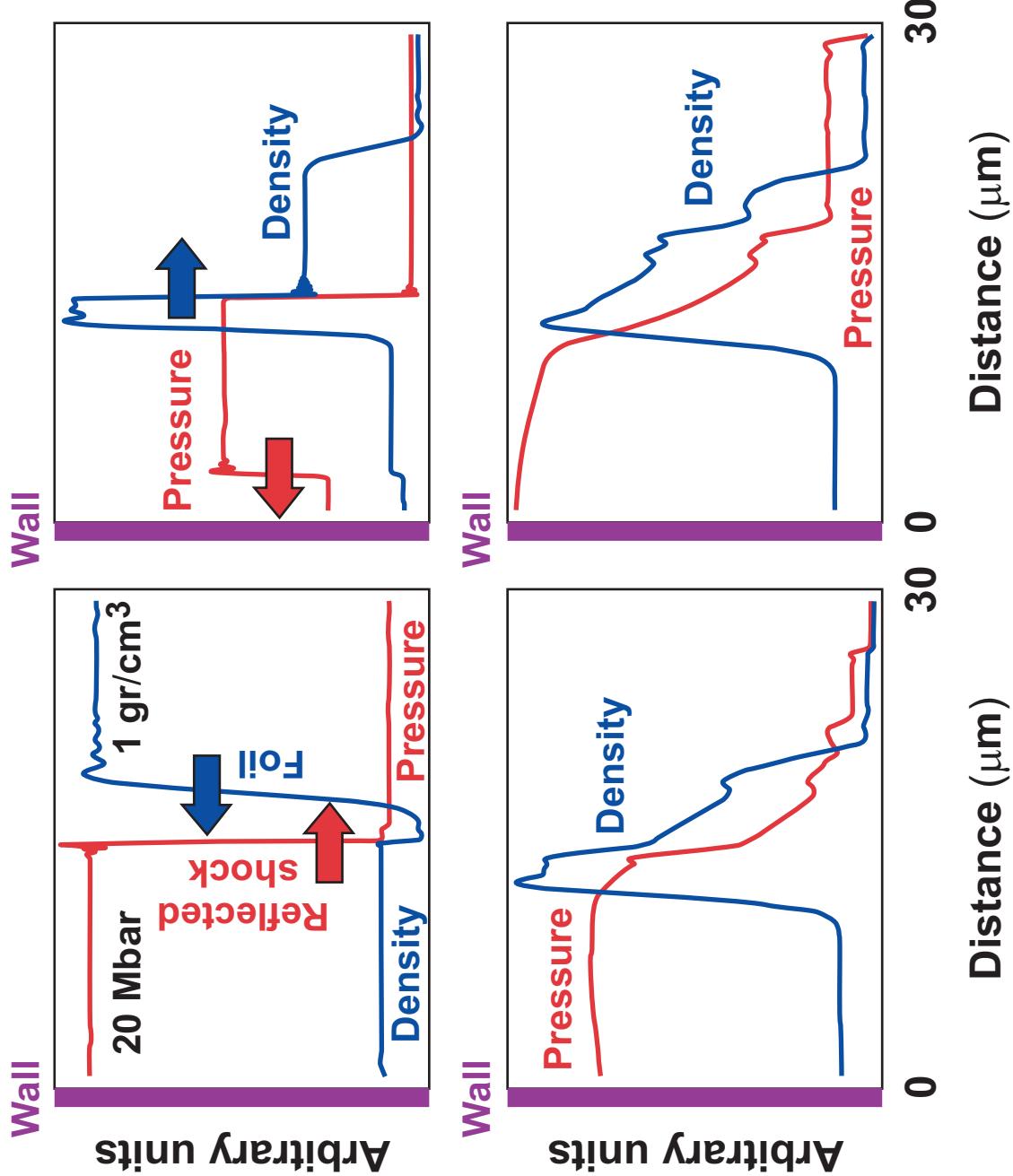


- The shock reflected from the wall slows down the foil, which in turn compresses the gas and decelerates.
- The 1-D problem can be solved analytically leading to a clear understanding of the relevant physics issues.

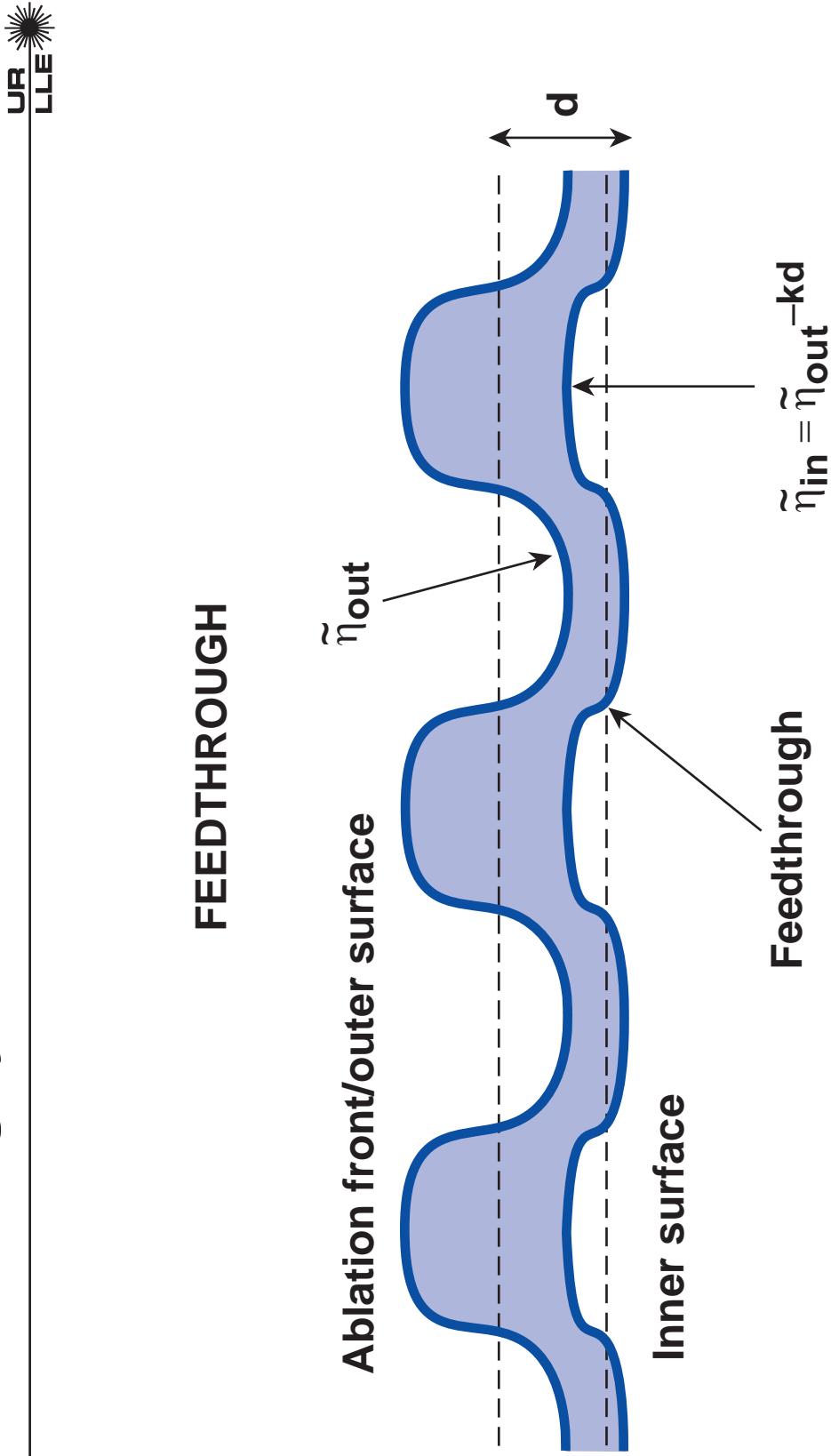
The foil is slowed down both impulsively and continuously



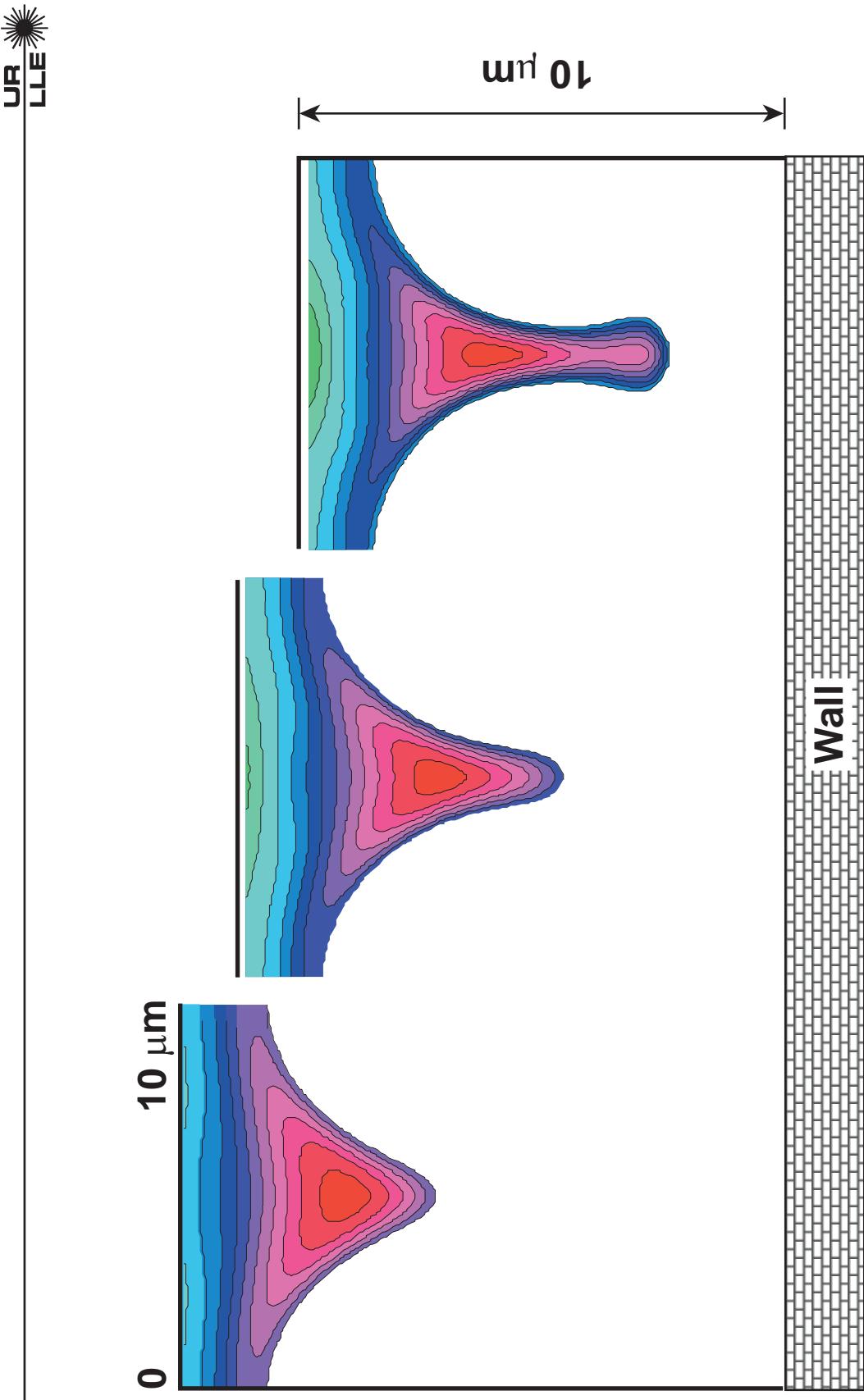
Evolution of pressure and density



Dec. RT seeding by outer-surface nonuniformities



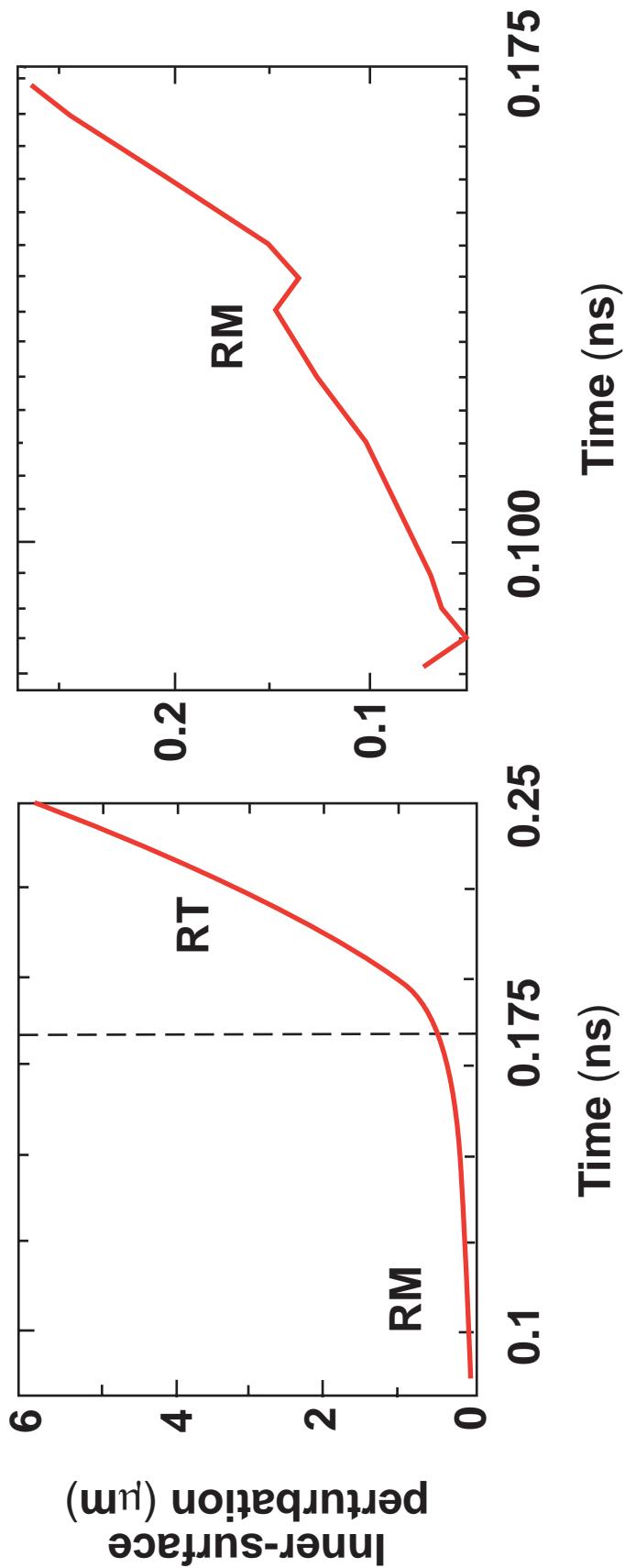
**As expected, the foil inner surface
is hydrodynamically unstable**



The deceleration-phase instability is a combination of Richtmyer–Meshkov and Rayleigh–Taylor



Linear growth of a 10- μm -wavelength perturbation

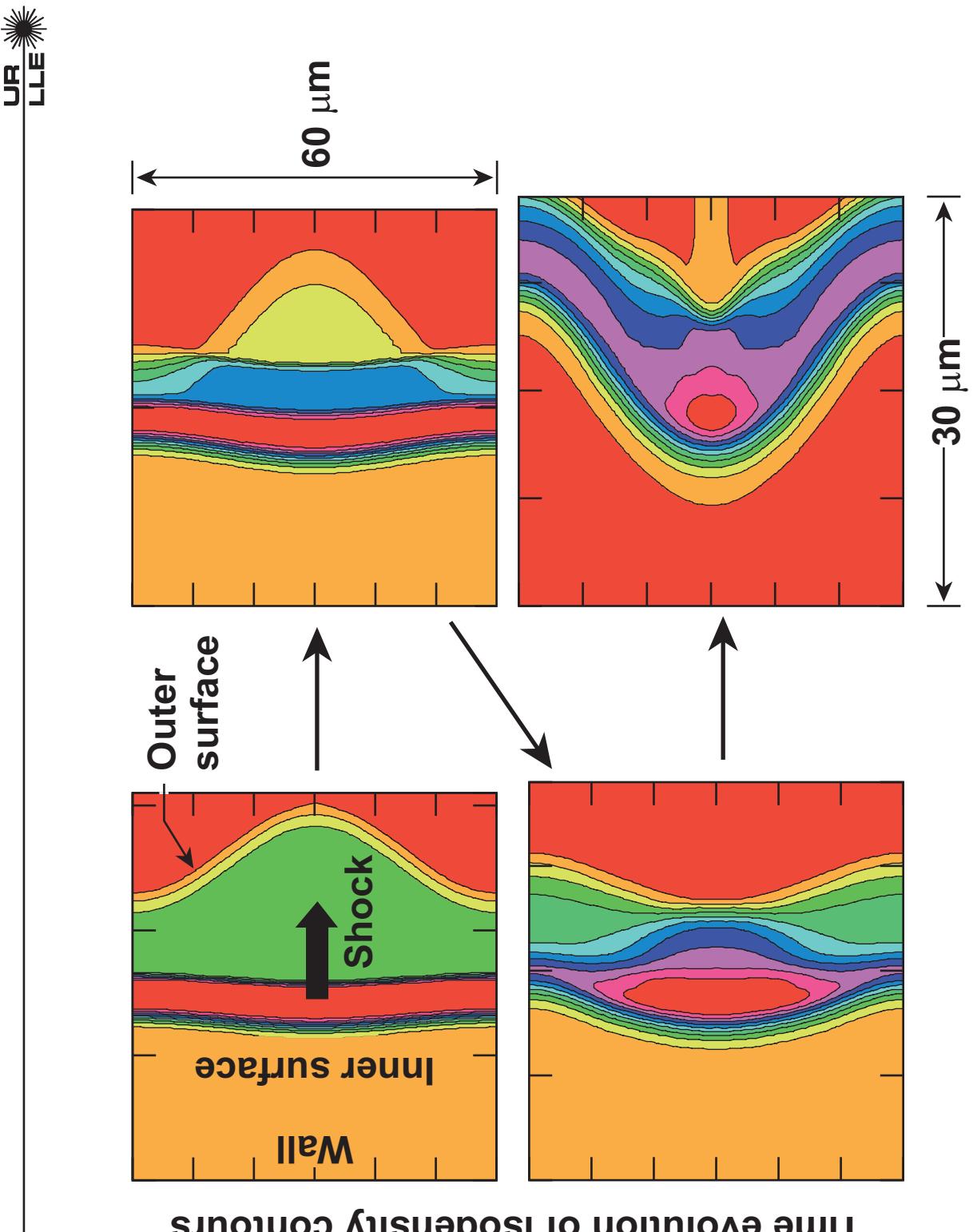


Dec. RT seeding by outer-surface nonuniformities

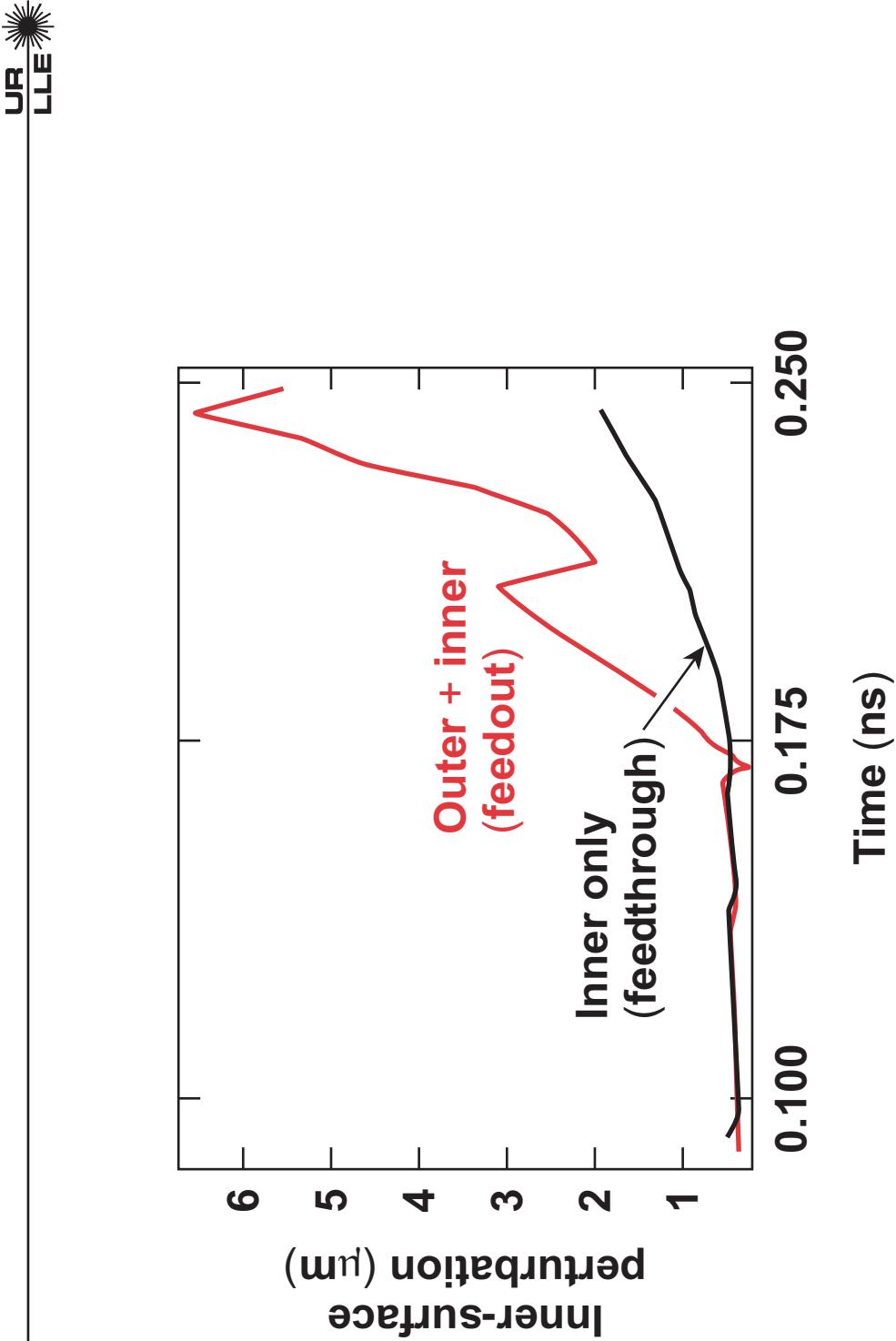


Reverse Feedout

**Reverse feedout is also a RT seed
during the deceleration phase**



The outer surface perturbations seed the deceleration RT



- The feedout is back!

Deceleration phase RT instability

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LLE

The Growth Rates

Not much is known about the RT growth rates during the deceleration phase

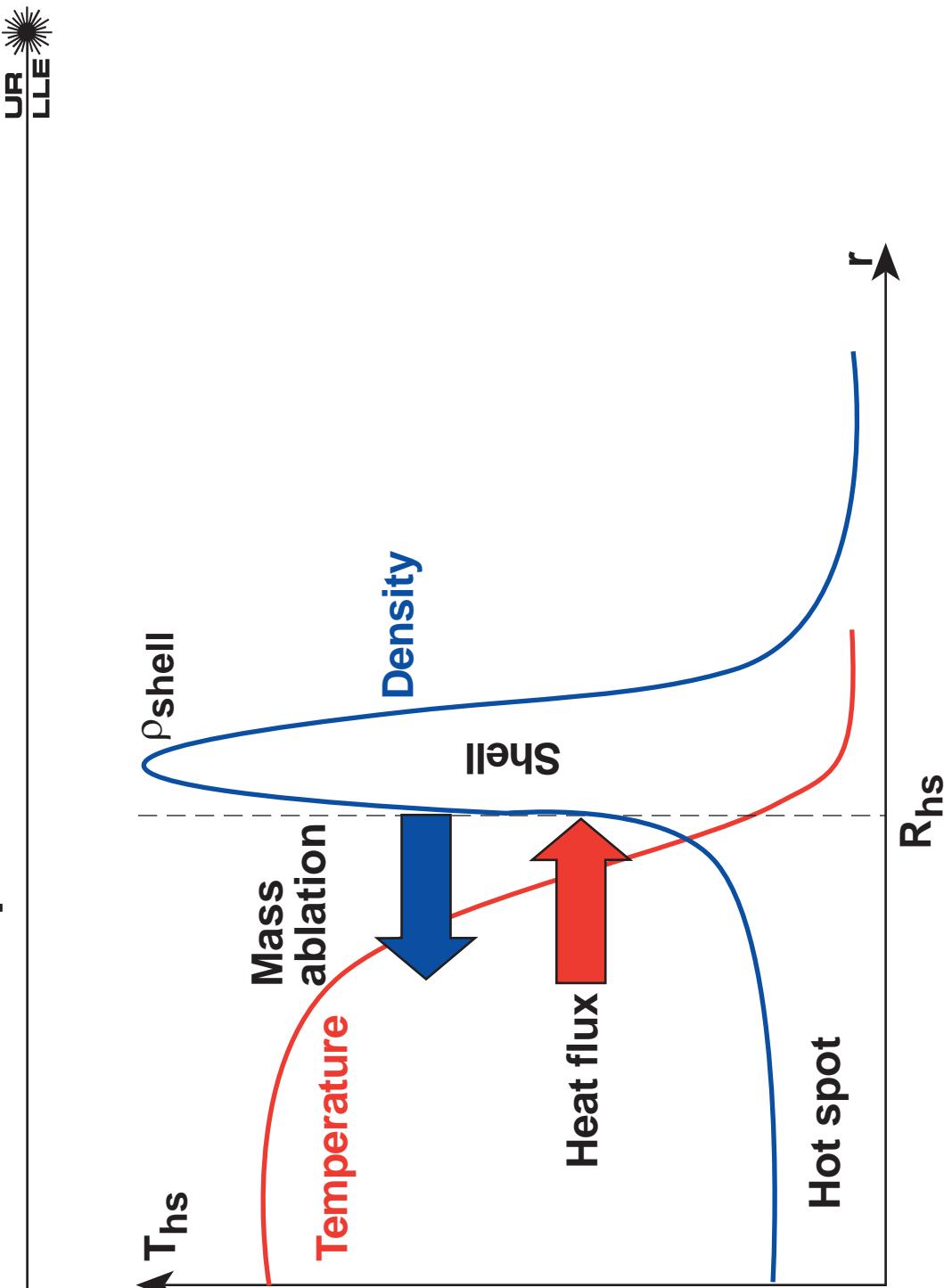


- The only mention of deceleration-phase RT growth rates is in Lindl's book:

$$\gamma_{\text{cl}} = \left[\frac{\text{kg}}{1 + \text{kL}} \right]^{1/2}$$

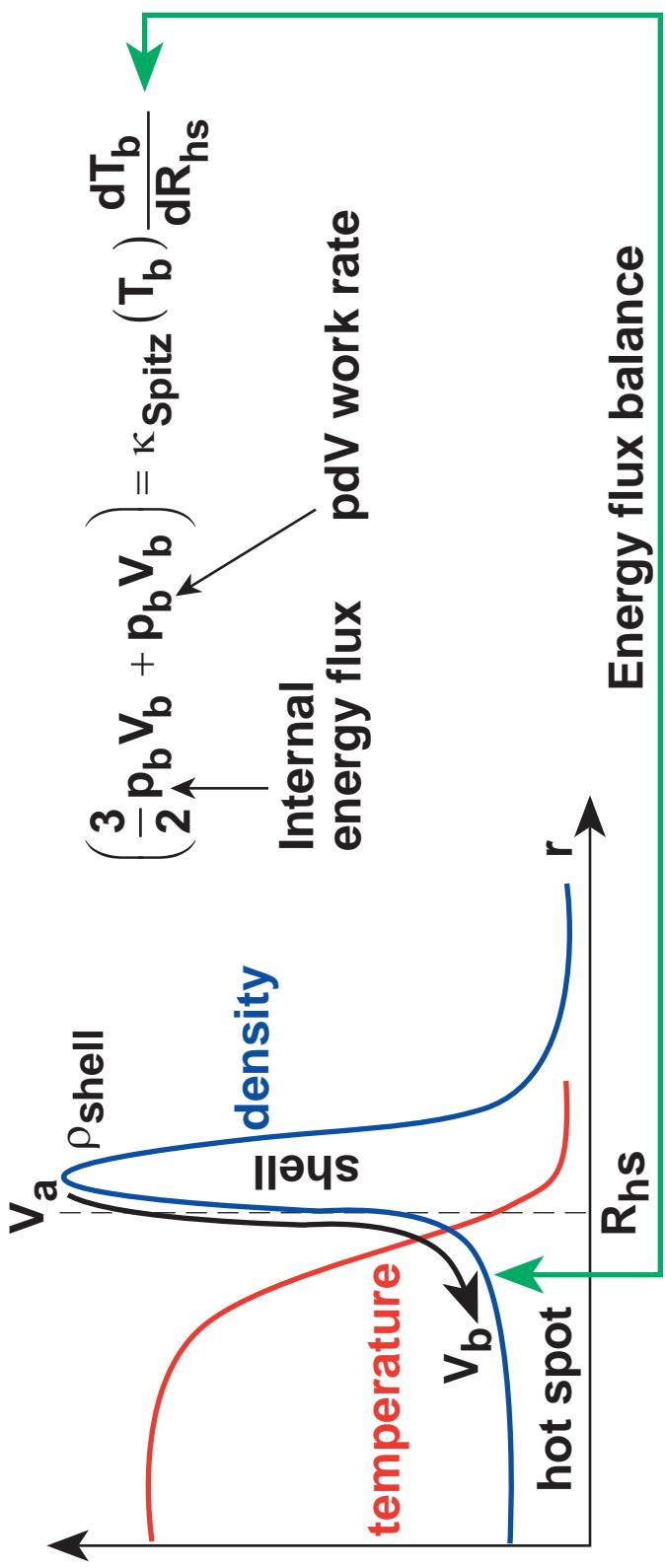
- Lindl's estimate for spherical implosions is $L \sim 0.2 R_{\text{hot}}$ spot.
- For NIF, R_{hot} spot ~ 50 to $100 \mu\text{m} \rightarrow L \sim 10$ to $20 \mu\text{m}$.
- Such a large L has a strong stabilizing effect.

The heat flux leaving the hot-spot is deposited on the shell surface causing mass ablation from the shell into the hot spot. The hot-spot mass increases in time



The ablation velocity is determined from the energy balance

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LLE



- Hot-spot temperature profile: $T_{hs} = T_0 \left(1 - \frac{r^2}{R_{hs}^2} \right)^{2/5}$

- Use the EOS: $p_b V_b = 2 \rho_b V_b T_b / M_i = 2 \dot{m} T_b / M_i$

- Ablation velocity: $V_a = \frac{\dot{m}}{\rho_{\text{shell}}} = 0.2 \frac{M_i \kappa_{\text{Spitz}} (T_0)}{\rho_{\text{shell}} R_{\text{hot spot}}}$

The density-gradient scale length is small



- Balance of heat flux to the shell and internal energy flux leaving the shell

$$\rho V_a \approx -\kappa(T_{sh}) \frac{dT_{sh}}{dr} \approx \kappa(T_{sh}) T_{sh} \frac{1}{\rho_{sh}} \frac{d\rho_{sh}}{dr}$$

- The density-gradient scale length is found using the formula for the ablation velocity:

$$L_m = \left[\frac{1}{\rho} \frac{d\rho}{dr} \right]_{min}^{-1} \approx 1.6 \frac{M_i \kappa(T_{sh})}{\rho_{sh} V_a} = 8 R_{hot-spot} \left(\frac{T_{shell}}{T_0} \right)^{5/2}$$

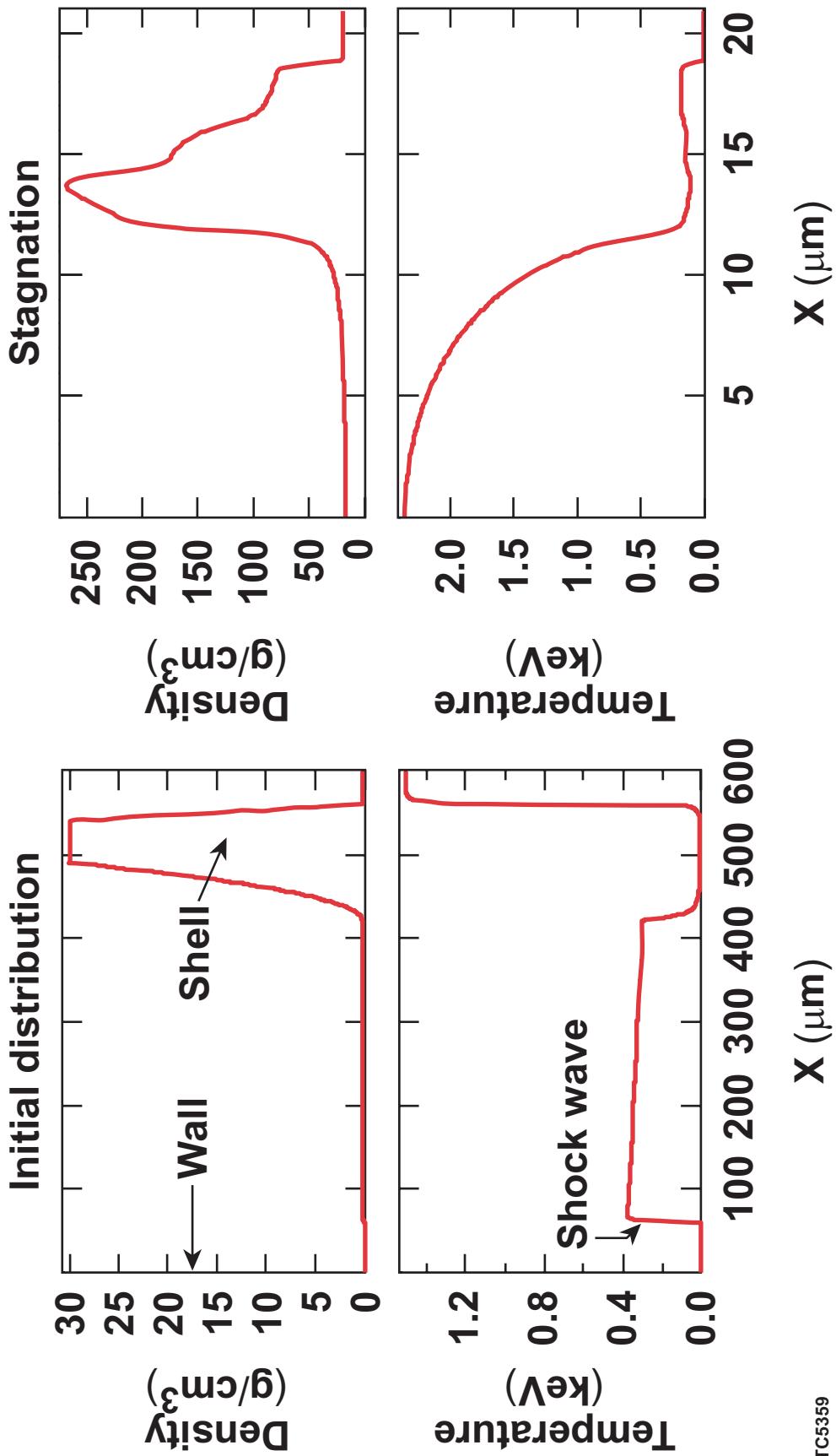
- For NIF: $L_m \sim 1 \mu m$

Planar Model

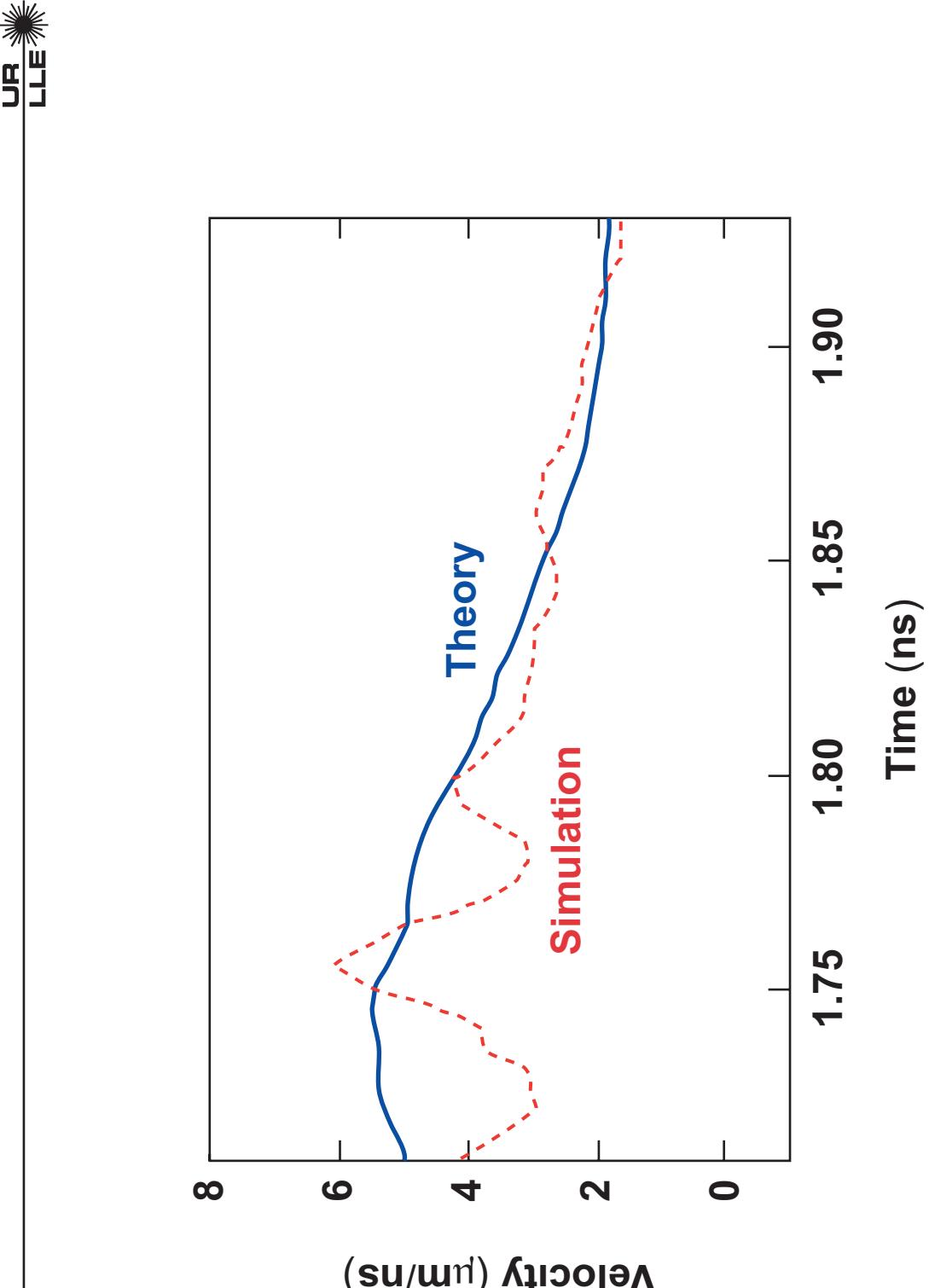
Planar simulations reproduce the behavior of ICF capsule implosions



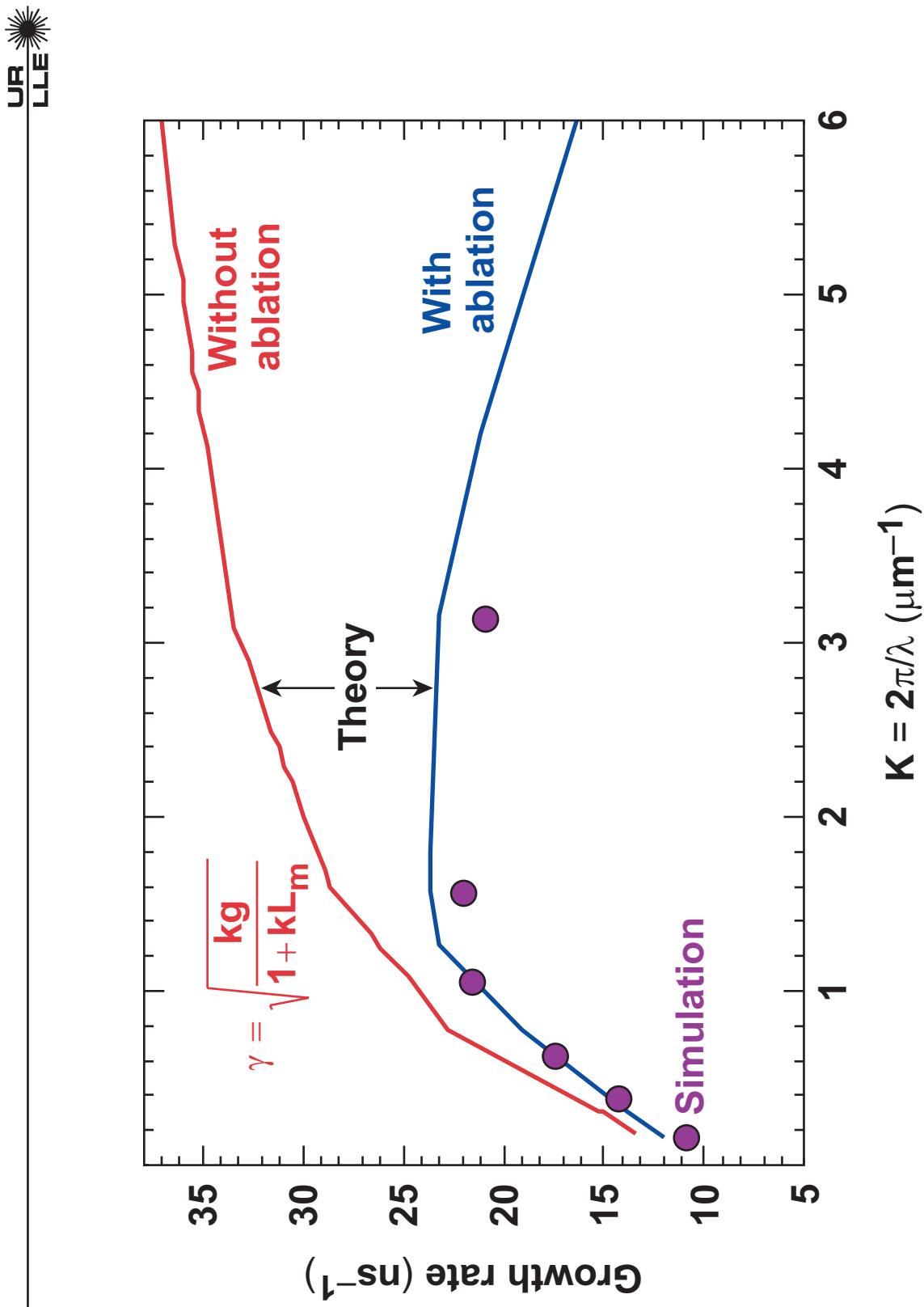
- Hot-spot temperature and radius and peak shell density have the same order of magnitude in planar and spherical cases.



Ablation velocity is significant
during the deceleration phase



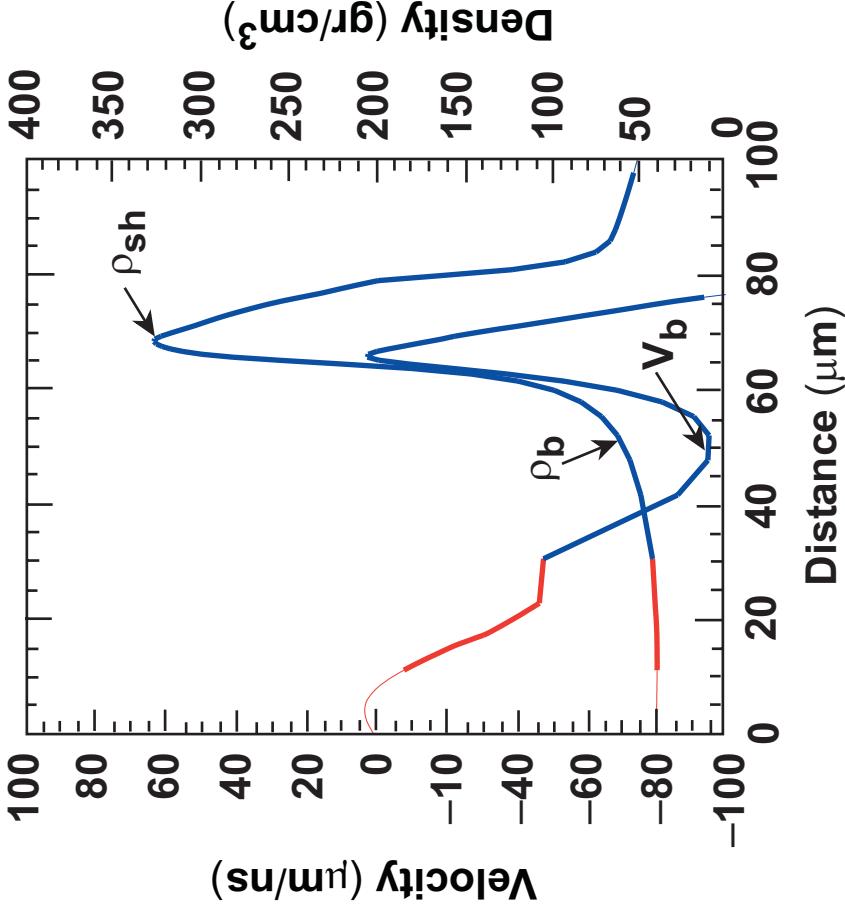
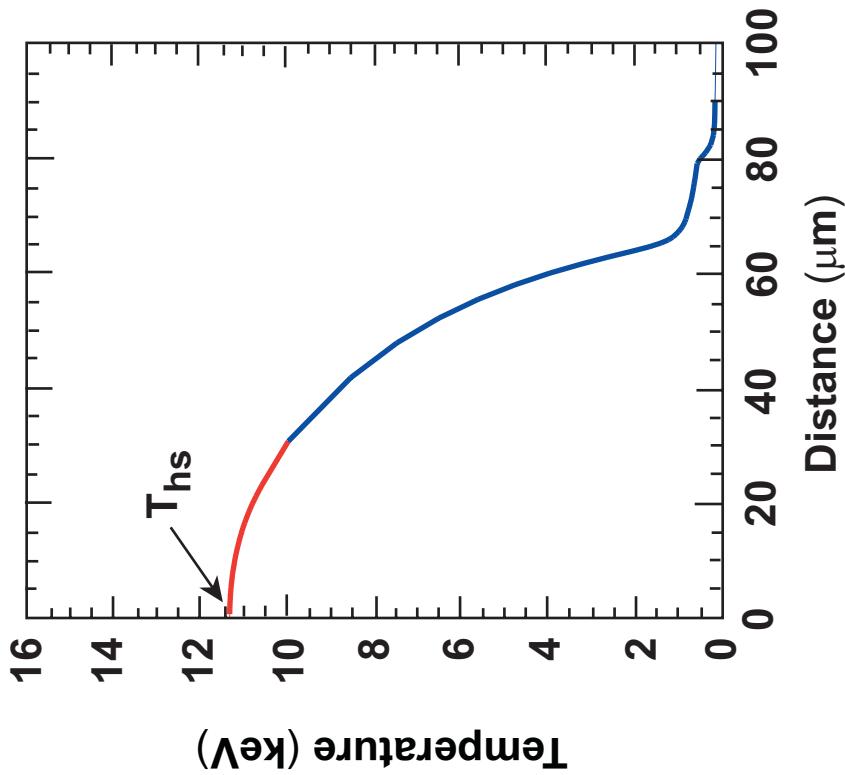
A significant reduction in RT growth rates is due to ablation



Theory and 1-D LILAC simulations yield the same value of the ablation velocity

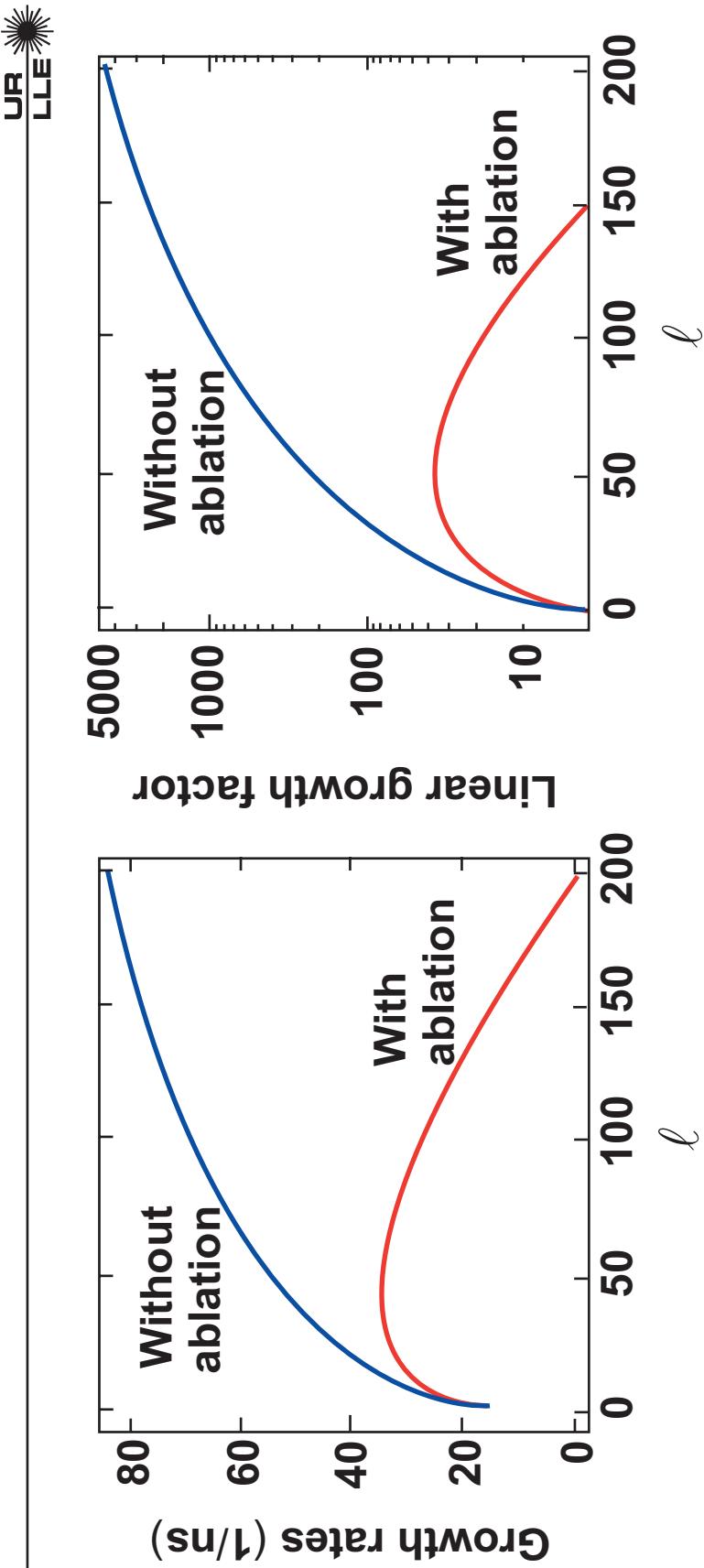


Time 10.40 ns



- From theory using $T_{hs} = 11.5 \text{ keV}$, $\rho_{sh} = 325 \text{ gr/cm}^3$, $R_{hs} = 65 \mu\text{m} \rightarrow V_a = 25 \mu\text{m/ns}$
- From simulations: $V_b = 100 \mu\text{m/ns}$, $\rho_b = 60 \text{ gr/cm}^3$, $\rho_b/\rho_{sh} = 20$; $V_a = \rho_b/V_b$

Theoretical NIF linear growth factors are significantly reduced by mass ablation



- Growth rate formula from R. Betti, et al., Phys. Plasma 1998.
- NIF deceleration phase: $g = 10^4 \text{ } \mu\text{m/ns}^2$, $V_a = 20 \text{ } \mu\text{m/ns}$, $R_{hs} = 70 \text{ } \mu\text{m}$, $L_m = 1 \text{ } \mu\text{m}$.
- Duration of deceleration phase $\sim 100 \text{ ps}$.

NIF cutoff: $\ell \approx 190$

Conclusions

- Two stages of the deceleration-phase instability are observed: deceleration by a series of shocks and continuous deceleration.
- Mass ablation through the inner surface and finite density-gradient scale length are the most important stabilizing effects.
- A significant reduction in the RT instability growth rate is due to the mass ablation.
- The inner-surface density-gradient scale length is lower than its standard estimate (~ 0.2 hot-spot radius).
- The cutoff wave number of the deceleration-phase RT for NIF is approximately ℓ cutoff = 150 to 200.